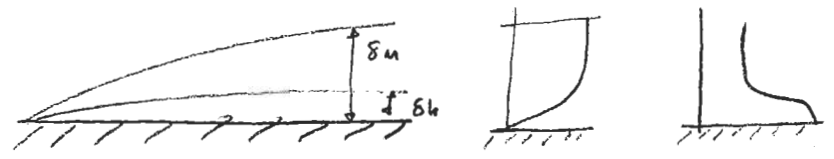


0 is



Pr > 1

SAE 30 Pr ≈ 3500.

for Pr > 1, $\frac{dp}{dx} = 0$

$$h_o = h_{ow} + (h_{oe} - h_{ow}) \frac{u}{u_e}$$

temp. profile given by

$$T = T_w + \left(T_e + \frac{u_e^2}{2c_p} - T_w \right) \frac{u}{u_e} - \frac{u^2}{2c_p}$$

If the wall is adiabatic $h_o = h_{ow} = \text{const}$

$$\therefore T_e + \frac{u_e^2}{2c_p} \equiv T_{aw} = \text{adiabatic wall temp}$$

$$\therefore T = T_w + (T_{aw} - T_w) \frac{u}{u_e} - \frac{u^2}{2c_p}$$

The wall ht flux is

$$q_w = k_w \frac{\partial T}{\partial y} \Big|_{\text{wall}} \quad \Big| \quad \frac{\partial T}{\partial y} = \dots$$

$$\therefore q_w = \frac{(T_{aw} - T_w) k_w u_w}{u_e \mu_w}$$

$$\text{or } C_h = \frac{q_w}{\rho_e u_e c_p (T_{aw} - T_w)} = \frac{C_f}{2Pr}$$

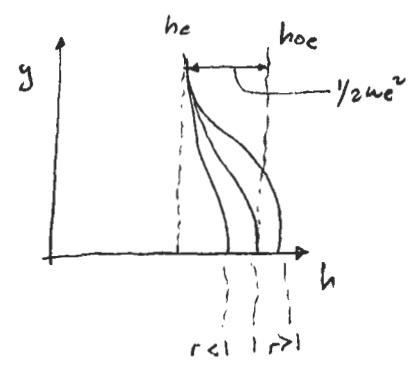
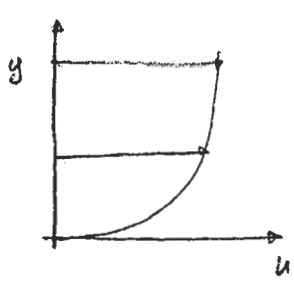
- Relates wall heat flux to the skin friction

- Reynolds analogy (effect valid)

- Strictly valid for Pr = 1 $\Rightarrow C_h = C_f/2$

(Stanton Number)
(wall ht transf coeff)

B) If $Pr \neq 1$ but close to unity as for gases; we can develop approximate wall temperature profile, which depends on Pr number.



Define temp recovery factor $r = f(Pr)$ where $f(1) = 1$

$$r = \frac{T_{aw} - T_e}{T_{oe} - T_e} \quad - \text{deviation from exact } Pr = 1 \text{ where } h_o(y) = \text{const}$$

Very good approximation for r

$$r \approx Pr^{1/2} \quad - \text{laminar}$$
$$\approx Pr^{1/3} \quad - \text{turbulent}$$

for air

$$r \approx 0.85 \quad \text{laminar}$$
$$\approx 0.9 \quad \text{turbulent}$$

Adiabatic wall:

$$h_{oa} = h_w |_{ad} = h_e + \frac{1}{2} r u_e^2 \quad (q_w = 0)$$

Non adiabatic: h_w prescribed

Note: That $h_o(y)$ matches exact profile only at $y=0$ & $y=\delta$ for $Pr \neq 1$

Given some $u(y)$ we can obtain density profile which is part of integral parameters δ^* , θ , etc.

Going back to Reynolds analogy.

(4)

$$C_h = \frac{C_f}{2Pr}$$

In general,

$$C_h = f/2 \cdot f(Pr)$$

empirical data shows

$$C_h \approx f/2 \cdot \frac{1}{Pr^{1/3}} \quad (\text{laminar and turbulent})$$

c) Integral BL Equ:

$$P/P_e \neq 1$$

Compressible VKI eqn:

$$\frac{d\theta}{dx} + (H+2 - M_e^2) \frac{\theta}{u_e} \frac{du_e}{dx} = f/2$$

Compressible K.E eqn:

$$\frac{1}{H^*} \frac{dH^*}{dx} = \frac{1}{\theta} \left(\frac{2C_p}{H^*} - f/2 \right) - \left(\frac{2H^{**}}{H^*} + 1 - H \right) \frac{1}{u_e} \frac{du_e}{dx}$$

$$\delta^* \equiv \int_0^\infty \left(1 - \frac{\rho u}{\rho_e u_e} \right) dy$$

$$\theta \equiv \int_0^\infty \left(1 - \frac{u}{u_e} \right) \frac{\rho u}{\rho_e u_e} dy$$

$$\theta^* \equiv \int_0^\infty \left(1 - \left(\frac{u}{u_e} \right)^2 \right) \frac{\rho u}{\rho_e u_e} dy$$

check \uparrow

$$\delta^{**} \equiv \int_0^\infty \left(1 - P/P_e \right) \frac{u}{u_e} dy$$

$O(M_e^2)$
for adiabatic flow

Note that

$$\frac{u_e}{\rho_e} \frac{d\rho_e}{du_e} = -M_e^2 \quad (\text{adiabatic freestream})$$

Shape parameter

$$H = \delta^* / \theta \quad (\text{compressible})$$

$$H^* = \theta^* / \theta$$

$$H^{**} = \delta^{**} / \theta \quad (\text{density} \quad \overset{\text{shape par}}{\cancel{\text{thickness}}})$$

↑
check

Compressible H and incompressible H_k are related (for adiab flow) in air

$$H_k = f(H, M_e^2)$$

$$H^{**} = f(H, M_e^2) \quad (\text{Whitfield})$$

See Data AAB paper.

Usual correlations apply with correction for density profile