16.06 Principles of Automatic Control Lecture 15

The Negative (0°) Root Locus

Sometimes, we need to plot the root locus for a negative gain parameter

Example:

Longitudinal dynamics of 747, M=0.8 at 20,000 ft. h =altitude δe =elevator deflection

$$\frac{h(s)}{\delta e(s)} = \frac{32.7(s+0.0045)(s+5.645)(s-5.61)}{s(s+0.003\pm0.0098j)(s+0.6463\pm1.1211j)}$$

Poles of system:

- s = 0 "energy mode" represents change in total (kinetic plus potential) energy. Hard to control with elevator hence near cancellation with zero at s = -0.0045.
- $s = -0.6463 \pm 1.1211 j$ "short period mode." The short period mode is dominated by changes in aircraft pitch altitude, much like an arrow or weather vane feathering into the wind.
- $s = -0.003 \pm 0.0098 j$ "phugoid mode". This mode represents long period exchange of kinetic and potential energy, with very small changes in aircraft angle of attack.

Note that:

- 1. There is a two orders of magnitude difference between the natural frequencies of the short period and phugoid modes.
- 2. The phugoid has only the modest damping ($\zeta = 0.29$). Often, it is much lower, say, only a few percent.

For our discussion, the most interesting aspect is the right half plane zero at s = +5.61. Why?

Rewrite the transfer function as

$$\frac{h(s)}{\delta e(s)} = G(s) = 32.7L(s)$$

$$\downarrow$$
root locus gain of G(s)

Look at the feedback loop using only proportional gain¹:



So the characteristic equation is:

$$0 = 1 + kG(s) = 1 + \underbrace{32.7k}_{K} L(s)$$

The question is, should k, (and therefore K) be positive or negative?

Note that because there are an off number of RHP poles and zeros, the D.C. gain G(0) is negative. To improve performance at D.C., must have *negative* gain.

There are other situations where a negative R.L. gain may be required, but this is the most common.

Modification to the Root Locus Rules.

The fundamental result is that s is on the negative locus if

$$1 + KL(s) = 0 \Rightarrow L(s) = -\frac{1}{K}$$

for some negative K. If K is negative, -1/K is positive, so that the angle condition is

¹For this system, a PD controller would be better, but that is not important for our argument.

$$\angle L(s) = 0^{\circ} + l \cdot 360^{\circ}, \quad linteger$$

So the rules are:

- <u>**Rule 1**</u>: The *n* branches of the locus start at the n poles. *m* approach the zeros, n-m approach ∞ . (No change)
- Rule 2: The locus is on the real axis to the left of an even number of poles and zeros.
- Rule 3: The asymptotes are described by

$$\begin{aligned} \alpha = & \frac{\sum p_i - \sum z_i}{n - m}, \quad \text{(no change)} \\ \theta_l = & \frac{360^\circ \cdot (l - 1)}{n - m}, \quad l = 1, 2, 3, \dots n - m \end{aligned}$$

Note: no 180° term.

• Rule 4: The departure angles from poles and the arrival angles at poles are given by

$$\begin{split} \phi_{\text{dep}} = & \frac{\sum \Psi_i - \sum^* \phi_i - 360^\circ \cdot (l-1)}{q} \\ \Psi_{\text{arr}} = & \frac{\sum \phi_i - \sum^* \Psi_i + 360^\circ \cdot (l-1)}{q} \end{split}$$

where q is multiplicity of pole or zero, l = 1, 2, ...q.

- **Rule 5:** The locus crosses the imaginary axis for values of K at which Routh's criterion shows a change in the number of unstable poles. (No change)
- Rule 6: No change in rule for when there are multiple points on the locus.

In summary, all rules are the same, except:

- 1. All $180^{\circ}s$ become $0^{\circ}s$.
- 2. "Odd" becomes "even" in Rule 1.

Example

$$G(s) = \frac{s-1}{(s+1)(s+3)}$$

 0° locus:



Note:

- Locus looks familiar but *is* different
- RHP zero tends to pull poles into RHP bad.

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