

## Application Tasks

5. The 2-D stress-strain equation for an orthotropic material can be written as:

$$\sigma_{\alpha\beta} = E_{\alpha\beta\gamma\delta}^* \epsilon_{\gamma\delta} \Rightarrow \begin{aligned} \sigma_{11} &= E_{1111}^* \epsilon_{11} + E_{1122}^* \epsilon_{22} \quad (1) \\ \sigma_{22} &= E_{1122}^* \epsilon_{11} + E_{2222}^* \epsilon_{22} \quad (2) \\ \sigma_{12} &= 2 E_{1212}^* \epsilon_{12} \quad (3) \end{aligned}$$

Next, write the stress-strain equations in terms of the strains (compliances! See Unit 5, p. 17)

$$\epsilon_{11} = \frac{1}{E_L} \sigma_{11} - \frac{\nu_{LT}}{E_L} \sigma_{22} - \frac{\nu_{L3}}{E_L} \sigma_{33} \quad (4)$$

$$\epsilon_{22} = -\frac{\nu_{TL}}{E_L} \sigma_{11} + \frac{1}{E_T} \sigma_{22} - \frac{\nu_{T3}}{E_T} \sigma_{33} \quad (5)$$

$$\epsilon_{33} = -\frac{\nu_{L3}}{E_L} \sigma_{11} - \frac{\nu_{T3}}{E_T} \sigma_{22} + \frac{1}{E_3} \sigma_{33} \quad (6)$$

$$\epsilon_{13} = \frac{1}{2G_{T3}} \sigma_{23} \quad (7)$$

$$\epsilon_{13} = \frac{1}{2G_{L3}} \sigma_{13} \quad (8)$$

$$\epsilon_{12} = \frac{1}{2G_{LT}} \sigma_{12} \quad (9)$$

\* Note that the subscripts L and T correspond to

1 and 2, respectively.

Now substitute the plane stress condition of

$$\sigma_{33} = \sigma_{13} = \sigma_{23} = 0 \quad (10)$$

into equations (4) - (9) to get:

$$(4) \rightarrow \epsilon_{11} = \frac{1}{E_L} \sigma_{11} - \frac{\nu_{LT}}{E_L} \sigma_{22} \quad (11)$$

$$(5) \rightarrow \epsilon_{22} = -\frac{\nu_{TL}}{E_T} \sigma_{11} + \frac{1}{E_T} \sigma_{22} \quad (12)$$

$$(6) \rightarrow \epsilon_{33} = -\frac{\nu_{L3}}{E_L} \sigma_{11} - \frac{\nu_{T3}}{E_T} \sigma_{22} \quad (13)$$

$$(7) \rightarrow \epsilon_{23} = 0 \quad (14)$$

$$(8) \rightarrow \epsilon_{13} = 0 \quad (15)$$

$$(9) \rightarrow \epsilon_{12} = \frac{1}{2G_{LT}} \sigma_{12} \quad (16)$$

The out-of-plane deformation is expressed in equation (13) but is not of primary interest in this problem. Neither are the out-of-plane shear strains which are zero as found in equations (14) and (15).

We can now relate the 2-D stiffness tensor,  $E_{ijkl}$ , to the elastic constants by comparing equation

(1) - (3) with equations (11), (12) and (16). Use matrix form to make the comparison straightforward.

For equations (1) - (3):

$$\begin{bmatrix} \sigma_{11} \\ \sigma_{22} \\ \sigma_{12} \end{bmatrix} = \begin{bmatrix} E_{1111}^* & E_{1122}^* & 0 \\ E_{1122}^* & E_{2222}^* & 0 \\ 0 & 0 & E_{1212}^* \end{bmatrix} \begin{bmatrix} \epsilon_{11} \\ \epsilon_{22} \\ \epsilon_{12} \end{bmatrix} \quad (17)$$

Similarly for (11), (12) and (16):

$$\begin{bmatrix} \epsilon_{11} \\ \epsilon_{22} \\ \epsilon_{12} \end{bmatrix} = \begin{bmatrix} 1/E_L & -\omega_L/E_L & 0 \\ -\omega_L/E_T & 1/E_T & 0 \\ 0 & 0 & 1/2G_{LT} \end{bmatrix} \begin{bmatrix} \sigma_{11} \\ \sigma_{22} \\ \sigma_{12} \end{bmatrix} \quad (18)$$

(2-D compliance form:  $\underline{\epsilon} = \underline{\sigma} \cdot \underline{\Sigma}$ )

In order to compare equation (17) with equation (18) one of the equations needs to be inverted.

Choosing equation (17):

$$\begin{bmatrix} \epsilon_{11} \\ \epsilon_{22} \\ \epsilon_{12} \end{bmatrix} = \begin{bmatrix} \bar{E}_{2222}^* & -E_{1122}^* & 0 \\ \frac{E_{1111}^* \bar{E}_{2222}^* - (E_{1122}^*)^2}{E_{1111}^* \bar{E}_{2222}^* - (E_{1122}^*)^2} & \frac{E_{1111}^* \bar{E}_{2222}^* - (E_{1122}^*)^2}{E_{1111}^* \bar{E}_{2222}^* - (E_{1122}^*)^2} & 0 \\ -\frac{E_{1122}^*}{E_{1111}^* \bar{E}_{2222}^* - (E_{1122}^*)^2} & \frac{E_{1111}^*}{E_{1111}^* \bar{E}_{2222}^* - (E_{1122}^*)^2} & 0 \\ 0 & 0 & \frac{1}{2\bar{E}_{1212}} \end{bmatrix} \begin{bmatrix} \sigma_{11} \\ \sigma_{22} \\ \sigma_{12} \end{bmatrix}$$

$\hookrightarrow (19)$

Now compare the elements of equations (18) and (19):

$$\bullet \quad \frac{1}{\bar{\epsilon}_L} = \frac{\bar{\epsilon}_{2222}^*}{\bar{\epsilon}_{1111}^* \bar{\epsilon}_{2222}^* - (\bar{\epsilon}_{1122}^*)^2}$$

$$\Rightarrow \bar{\epsilon}_L = \bar{\epsilon}_{1111}^* - \frac{(\bar{\epsilon}_{1122}^*)^2}{\bar{\epsilon}_{2222}^*} \quad (20a)$$

$$\bullet \quad -\frac{\omega_{LT}}{\bar{\epsilon}_L} = -\frac{\bar{\epsilon}_{1122}^*}{\bar{\epsilon}_{1111}^* \bar{\epsilon}_{2222}^* - (\bar{\epsilon}_{1122}^*)^2}$$

$$\Rightarrow \omega_{LT} = \frac{\bar{\epsilon}_{1122}^*}{\bar{\epsilon}_{1111}^* \bar{\epsilon}_{2222}^* - (\bar{\epsilon}_{1122}^*)^2} \bar{\epsilon}_L$$

Using the result above of equation (20a) :

$$\omega_{LT} = \frac{\bar{\epsilon}_{1122}^*}{\bar{\epsilon}_{1111}^* \bar{\epsilon}_{2222}^* - (\bar{\epsilon}_{1122}^*)^2} \frac{\bar{\epsilon}_{1111}^* \bar{\epsilon}_{2222}^* - (\bar{\epsilon}_{1122}^*)^2}{\bar{\epsilon}_{2222}^*}$$

$$\Rightarrow \omega_{LT} = \frac{\bar{\epsilon}_{1122}^*}{\bar{\epsilon}_{2222}^*} \quad (20b)$$

$$\bullet \quad \frac{1}{\bar{\epsilon}_T} = \frac{\bar{\epsilon}_{1111}^*}{\bar{\epsilon}_{1111}^* \bar{\epsilon}_{2222}^* - (\bar{\epsilon}_{1122}^*)^2}$$

$$\Rightarrow \bar{\epsilon}_T = \bar{\epsilon}_{2222}^* - \frac{(\bar{\epsilon}_{1122}^*)^2}{\bar{\epsilon}_{1111}^*} \quad (20c)$$

$$\bullet \quad -\frac{\omega_{TL}}{\bar{\epsilon}_T} = -\frac{\bar{\epsilon}_{1122}^*}{\bar{\epsilon}_{1111}^* \bar{\epsilon}_{2222}^* - (\bar{\epsilon}_{1122}^*)^2}$$

using the result above of equation (20c):

$$\omega_{TC} = \frac{E_{1122}^*}{E_{1111}^* - (\bar{E}_{2222}^*)^2} \cdot \frac{\bar{E}_{1111}^* \bar{E}_{2222}^* - (E_{1122}^*)^2}{\bar{E}_{1111}^*}$$

$$\Rightarrow \omega_{TC} = \frac{E_{1122}^*}{E_{1111}^*} \quad (20d)$$

$$\bullet \quad \frac{1}{2G_{LT}} = \frac{1}{2\bar{E}_{1212}^*}$$

$$\Rightarrow G_{LT} = \bar{E}_{1212}^* \quad (20e)$$

Summarizing:

$$E_L = E_{1111}^* - \frac{(E_{1122}^*)^2}{\bar{E}_{2222}^*}$$

$$E_T = \bar{E}_{2222}^* - \frac{(E_{1122}^*)^2}{\bar{E}_{1111}^*}$$

$$\omega_{TC} = \frac{E_{1122}^*}{\bar{E}_{2222}^*}$$

(20a-e)

$$\omega_{TC} = \frac{E_{1122}^*}{E_{1111}^*}$$

$$G_{LT} = \bar{E}_{1212}^*$$

If instead of inverting equation (17) one inverted equation (18) and then did the comparison of equations one would get the relations in inverse:  $\bar{E}_{\text{ext}}^*$  in terms of the in-plane engineering constant:

$$\boxed{\begin{aligned}\bar{E}_{1111}^* &= \frac{E_L}{1 - \nu_{LT} \nu_{TL}} \\ \bar{E}_{2222}^* &= \frac{\nu_{LT} \bar{E}_T}{1 - \nu_{LT} \nu_{TL}} = \frac{\nu_{TL} E_L}{1 - \nu_{LT} \nu_{TL}} \\ \bar{E}_{2222}^* &= \frac{E_T}{1 - \nu_{LT} \nu_{TL}} \\ \bar{E}_{1212}^* &= G_{LT}\end{aligned}} \quad (21)$$

6. For the isotropic case, the same relation between  $\bar{E}_{\text{ext}}^*$  and the elastic constant hold. The differences are that isotropic materials have the same properties in all directions:

$$\bar{E}_L = \bar{E}_T = \bar{E}$$

$$\nu_{LT} = \nu_{TL} = \nu$$

$$\text{and } \bar{E}_{1111}^* = \bar{E}_{2222}^* \text{ (found in problem #2)}$$

and there is a relationship between  $E_N$  and  $G_I$ :

$$G = \frac{E}{2(1+n)}$$

We also had from Problem #2:

$$E_{1212}^* = \frac{E_{1111} - E_{1122}}{2}$$

with:

$$E_{1111}^* = E_{1111} - \frac{E_{1122}^2}{E_{1111}} \quad \text{and} \quad E_{1122}^* = E_{1122} - \frac{E_{1122}^2}{E_{1111}}$$

so:

$$E_{1111} = E_{1111}^* + \frac{E_{1122}^2}{E_{1111}} \quad \text{and} \quad E_{1122} = E_{1122}^* + \frac{E_{1122}^2}{E_{1111}}$$

using in the expression for  $\bar{E}_{1212}^+$ :

$$\bar{E}_{1212}^+ = \frac{1}{2} \left\{ \bar{E}_{1111}^* + \frac{E_{1122}^2}{E_{1111}} - \left( E_{1122}^* + \frac{E_{1122}^2}{E_{1111}} \right) \right\}$$

$$\text{giving: } E_{1212}^* = \frac{1}{2} (E_{1111}^* - E_{1122}^*)$$

again, only 2 independent constants

using these facts in equations (20 a-e)

|  |   |
|--|---|
| $E_L = \bar{E} = \bar{E}_{1111}^* - \frac{(E_{1122}^*)^2}{E_{1111}^*}$ | <span style="font-size: 2em;">\</span> the same |
| $E_T = E = E_{1111}^* - \frac{(E_{1122}^*)^2}{E_{1111}^*}$             | <span style="font-size: 2em;">\</span>          |
| <span style="font-size: 2em;">:</span>                                 | <span style="font-size: 2em;">:</span>          |

$$\boxed{\begin{array}{l} N_{LT} = \frac{\epsilon_{1122}^*}{\epsilon_{1111}^*} \\ N_{TC} = \frac{\epsilon_{1122}^*}{\epsilon_{1111}^*} \\ G_{LT} = \frac{\epsilon_{1111}^* - \epsilon_{1122}^*}{2} \end{array}} \quad \text{the same}$$

If one goes the other way and uses equations (21), then:

$$\boxed{\begin{array}{l} \epsilon_{1111}^* = \frac{\epsilon}{1-\omega^2} \\ \epsilon_{1122}^* = \frac{\omega \epsilon}{1-\omega^2} \\ \epsilon_{2222}^* = \frac{\epsilon}{1-\omega^2} \\ \epsilon_{1212}^* = G = \frac{\epsilon}{2(1+\omega)} \end{array}} \quad \leftarrow \text{the same or } \epsilon_{1111}^*$$

Does  $\epsilon_{1212}^* : \frac{1}{2}(\epsilon_{1111}^* - \epsilon_{1122}^*)$ ? check ....

$$\epsilon_{1212}^* = \frac{\epsilon}{2(1+\omega)} \stackrel{?}{=} \frac{1}{2}(\epsilon_{1111}^* - \epsilon_{1122}^*) = \frac{\epsilon - \omega \epsilon}{2(1+\omega)}$$

$$1-\omega^2 = (1+\omega)(1-\omega)$$

$$\Rightarrow \frac{\epsilon - nE}{2(1-n)^2} : \frac{\epsilon(1-n)}{2(r+n)(r-n)}$$

$$f_0 : \frac{\epsilon}{2(1+n)} \stackrel{?}{=} \frac{\epsilon}{2(1+n)} \quad \underline{\underline{Y = S'}}$$