

# 2.43 ADVANCED THERMODYNAMICS

**Spring Term 2024**

**LECTURE 18**

Room 3-442

Friday, April 12, 11:00am - 1:00pm

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Room 3-351d

# **Systems with chemical reactions**

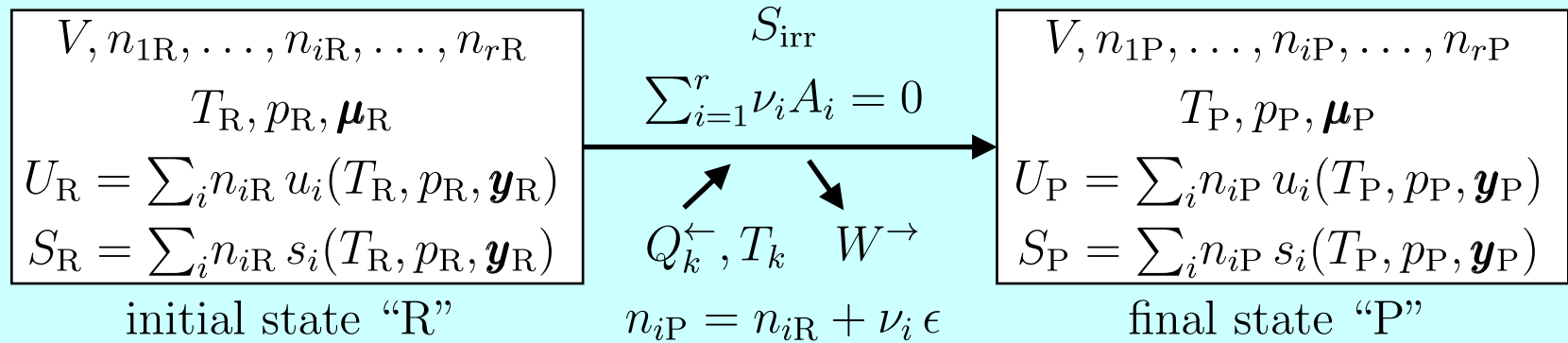
**energy and entropy balances**

**notation and stoichiometry**

**properties of reaction**

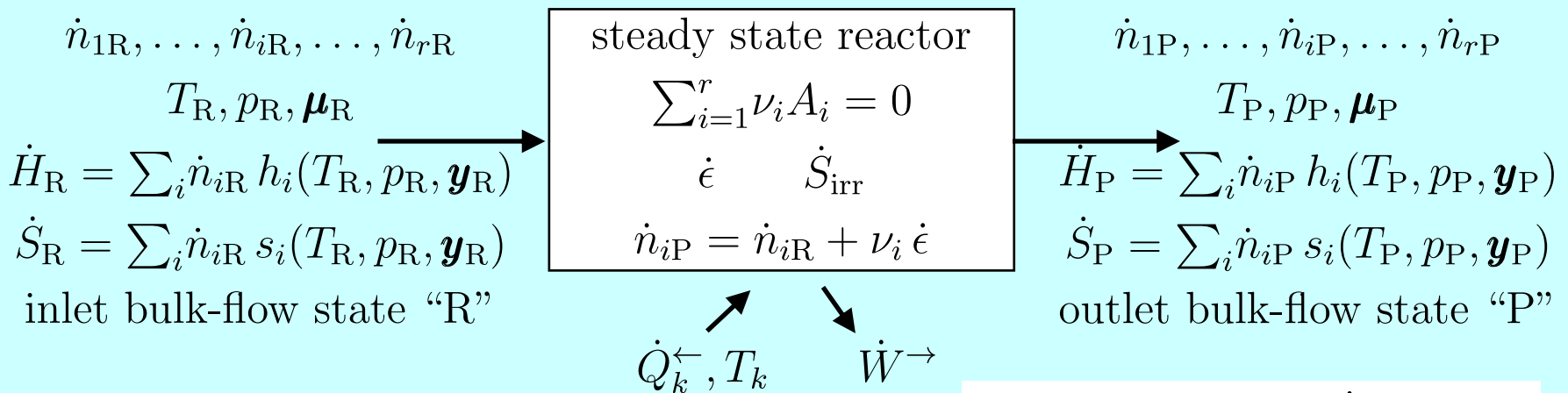
**properties of formation**

# Energy and entropy balances for closed and open systems with chemical reactions



$$U_P - U_R = \sum_k Q_k^{\leftarrow} - W^{\rightarrow}$$

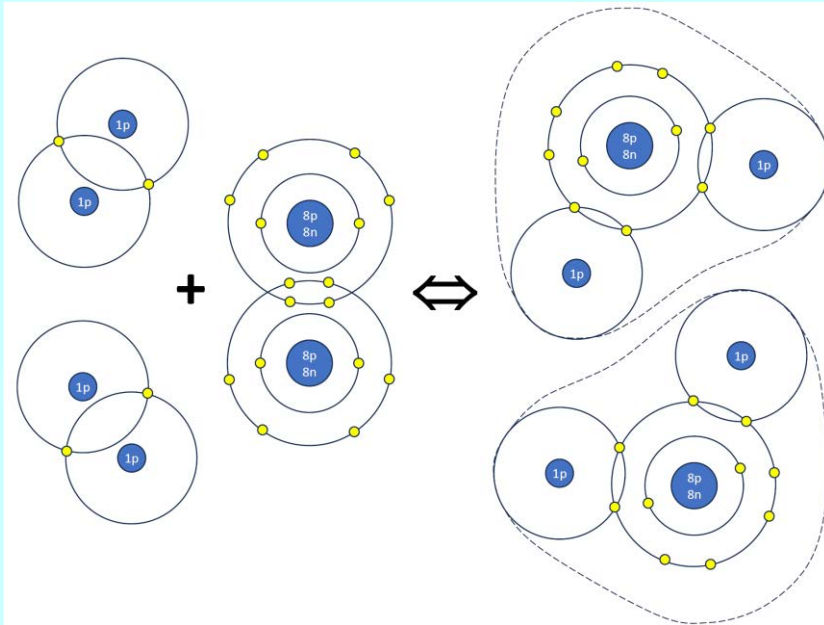
$$S_P - S_R = \sum_k \frac{Q_k^{\leftarrow}}{T_k} + S_{\text{irr}}$$



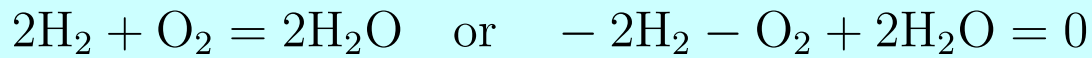
$$0 = \dot{H}_R - \dot{H}_P + \sum_k \dot{Q}_k^{\leftarrow} - \dot{W}^{\rightarrow}$$

$$0 = \dot{S}_R - \dot{S}_P + \sum_k \frac{\dot{Q}_k^{\leftarrow}}{T_k} + \dot{S}_{\text{irr}}$$

# proportionality relations, reaction coordinates



chemical reaction mechanism



composition:  $n_{\text{H}_2}, n_{\text{O}_2}, n_{\text{H}_2\text{O}}$

changes in composition:  $\Delta n_{\text{H}_2}, \Delta n_{\text{O}_2}, \Delta n_{\text{H}_2\text{O}}$

proportionality relations

$$\frac{\Delta n_{\text{H}_2}}{-2} = \frac{\Delta n_{\text{O}_2}}{-1} = \frac{\Delta n_{\text{H}_2\text{O}}}{2} = \Delta \epsilon$$

$\epsilon =$  reaction coordinate

chemical symbols

$$A_1 = \text{H}_2, \quad A_2 = \text{O}_2, \quad A_3 = \text{H}_2\text{O}$$

stoichiometric coefficients

$$\nu_1 = -2, \quad \nu_2 = -1, \quad \nu_3 = 2$$

rewrite the mechanism as

$$\nu_1 A_1 + \nu_2 A_2 + \nu_3 A_3 = 0$$

$$\sum_i \nu_i A_i = 0$$

proportionality relations

$$\frac{\Delta n_1}{\nu_1} = \frac{\Delta n_2}{\nu_2} = \frac{\Delta n_3}{\nu_3} = \Delta \epsilon$$

$$\Delta n_i = \nu_i \Delta \epsilon \quad \forall i$$

$$n_{i\text{P}} = n_{i\text{R}} + \nu_i \epsilon$$

where  $\epsilon = \epsilon_{\text{P}} - \epsilon_{\text{R}}$

$$\dot{n}_i = \nu_i \dot{\epsilon} \quad \forall i$$

$$\dot{n}_{i\text{P}} = \dot{n}_{i\text{R}} + \nu_i \dot{\epsilon}$$

# Properties of reaction

The balance equations require these differences:  $U_P - U_R$ ,  $S_P - S_R$ ,  $\dot{H}_P - \dot{H}_R$ ,  $\dot{S}_P - \dot{S}_R$ . For example,

$$\begin{aligned}
 U_P - U_R &= \sum_i n_{iP} u_i(T_P, p_P, y_{iP}) - \sum_i n_{iR} u_i(T_R, p_R, y_{iR}) = \text{add and subtract } u_{ii}(T, p_o) \\
 &= \sum_i n_{iP} \underbrace{[u_i(T_P, p_P, y_{iP}) - u_{ii}(T, p_o)]}_{\substack{\Delta u_i^{\text{mix}}|_{T_P, p_P} + \\ + u_{ii}(T_P, p_P) - u_{ii}(T, p_o)}} - \sum_i n_{iR} \underbrace{[u_i(T_R, p_R, y_{iR}) - u_{ii}(T, p_o)]}_{\substack{\Delta u_i^{\text{mix}}|_{T_R, p_R} + \\ + u_{ii}(T_R, p_R) - u_{ii}(T, p_o)}} + \underbrace{\sum_i (n_{iP} - n_{iR}) u_{ii}(T, p_o)}_{\substack{\nu_i \epsilon \\ \epsilon \sum_i \nu_i u_{ii}(T, p_o)}} = \epsilon \Delta u^o(T)
 \end{aligned}$$

$$\begin{aligned}
 \dot{H}_P - \dot{H}_R &= \sum_i \dot{n}_{iP} h_i(T_P, p_P, y_{iP}) - \sum_i \dot{n}_{iR} h_i(T_R, p_R, y_{iR}) = \text{add and subtract } h_{ii}(T, p_o) \\
 &= \sum_i \dot{n}_{iP} \underbrace{[h_i(T_P, p_P, y_{iP}) - h_{ii}(T, p_o)]}_{\substack{\Delta h_i^{\text{mix}}|_{T_P, p_P} + \\ + h_{ii}(T_P, p_P) - h_{ii}(T, p_o)}} - \sum_i \dot{n}_{iR} \underbrace{[h_i(T_R, p_R, y_{iR}) - h_{ii}(T, p_o)]}_{\substack{\Delta h_i^{\text{mix}}|_{T_R, p_R} + \\ + h_{ii}(T_R, p_R) - h_{ii}(T, p_o)}} + \underbrace{\sum_i (\dot{n}_{iP} - \dot{n}_{iR}) h_{ii}(T, p_o)}_{\substack{\nu_i \dot{\epsilon} \\ \dot{\epsilon} \sum_i \nu_i h_{ii}(T, p_o)}} = \dot{\epsilon} \Delta h^o(T)
 \end{aligned}$$

So, we defined the properties of reaction

at  $T$  and standard pressure

$p_o = 1$  atm:

$$\begin{aligned}
 \Delta u^o(T) &= \sum_i \nu_i u_{ii}(T, p_o) \\
 \Delta s^o(T) &= \sum_i \nu_i s_{ii}(T, p_o) \\
 \Delta h^o(T) &= \sum_i \nu_i h_{ii}(T, p_o) \\
 \Delta g^o(T) &= \sum_i \nu_i g_{ii}(T, p_o) \\
 \Delta g^o(T) &= \Delta h^o(T) - T \Delta s^o(T) \\
 \Delta v^o(T) &= \sum_i \nu_i v_{ii}(T, p_o) \\
 \Delta h^o(T) &= \Delta u^o(T) + p_o \Delta v^o(T)
 \end{aligned}$$

at STP:  $T_o = 298.15$  K

and  $p_o = 1$  atm:

$$\begin{aligned}
 \Delta u^o &= \sum_i \nu_i u_{ii}(T_o, p_o) \\
 \Delta s^o &= \sum_i \nu_i s_{ii}(T_o, p_o) \\
 \Delta h^o &= \sum_i \nu_i h_{ii}(T_o, p_o) \\
 \Delta g^o &= \sum_i \nu_i g_{ii}(T_o, p_o) \\
 \Delta g^o &= \Delta h^o - T_o \Delta s^o \\
 \Delta v^o &= \sum_i \nu_i v_{ii}(T_o, p_o) \\
 \Delta h^o &= \Delta u^o + p_o \Delta v^o
 \end{aligned}$$

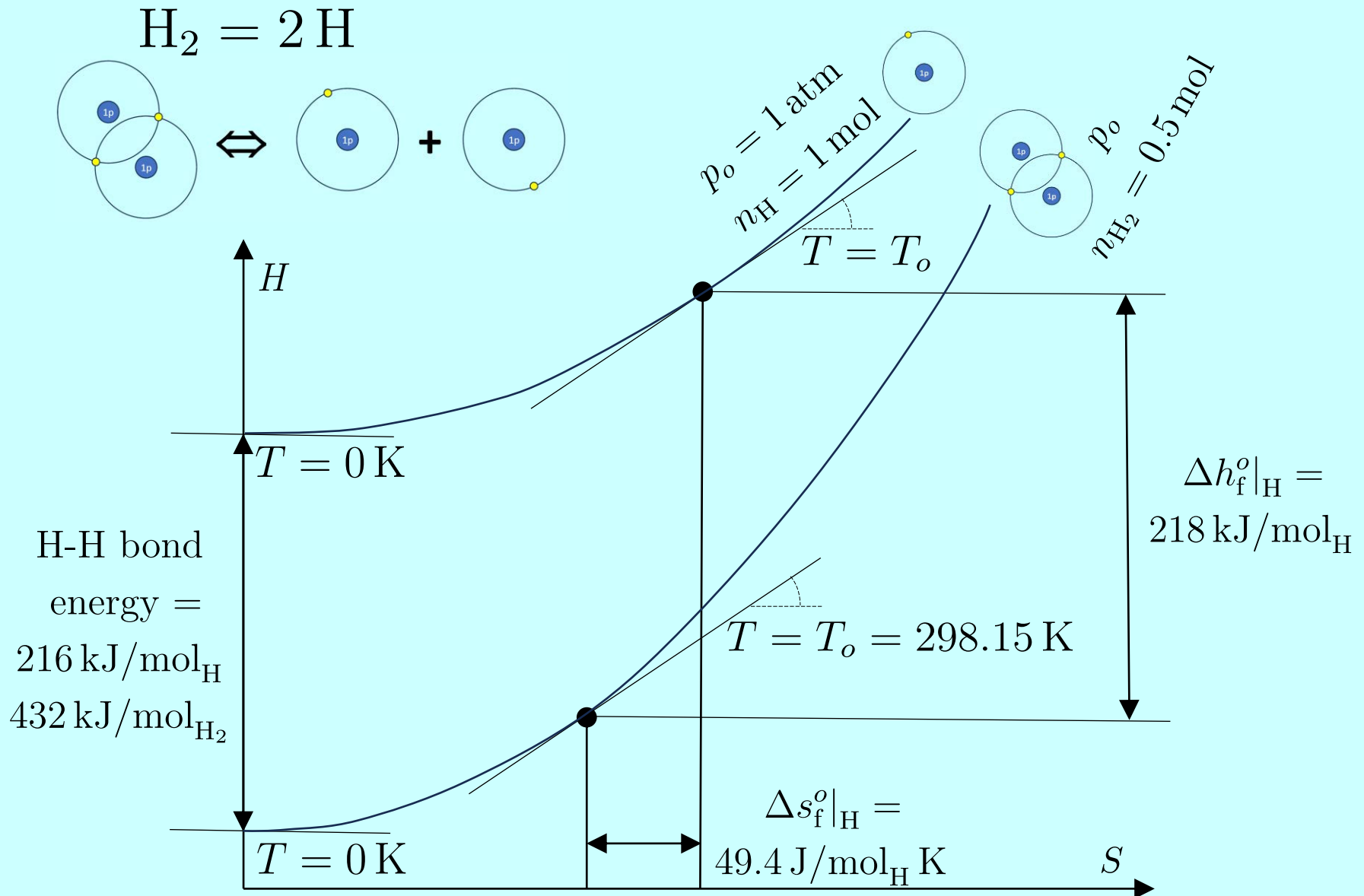
using the Hess relation

and the properties of

formation at STP:

$$\begin{aligned}
 \Delta u^o &= \sum_i \nu_i \Delta u_f^o|_i \\
 \Delta s^o &= \sum_i \nu_i \Delta s_f^o|_i \\
 \Delta h^o &= \sum_i \nu_i \Delta h_f^o|_i \\
 \Delta g^o &= \sum_i \nu_i \Delta g_f^o|_i \\
 \Delta v^o &= \sum_i \nu_i \Delta v_f^o|_i
 \end{aligned}$$

# Enthalpy of formation of H and H-H bond energy



# van der Waals forces vs covalent bonds

van der Waals forces have a significant impact on the physical properties of substances, such as boiling points, melting points, and solubility. But they are very weak compared to ionic or covalent bonds.

$$\epsilon = 459 \text{ kJ/mol}$$

$$r_{\min} = 2^{1/6} \sigma = 0.74 \text{ \AA} = 74 \text{ pm}$$

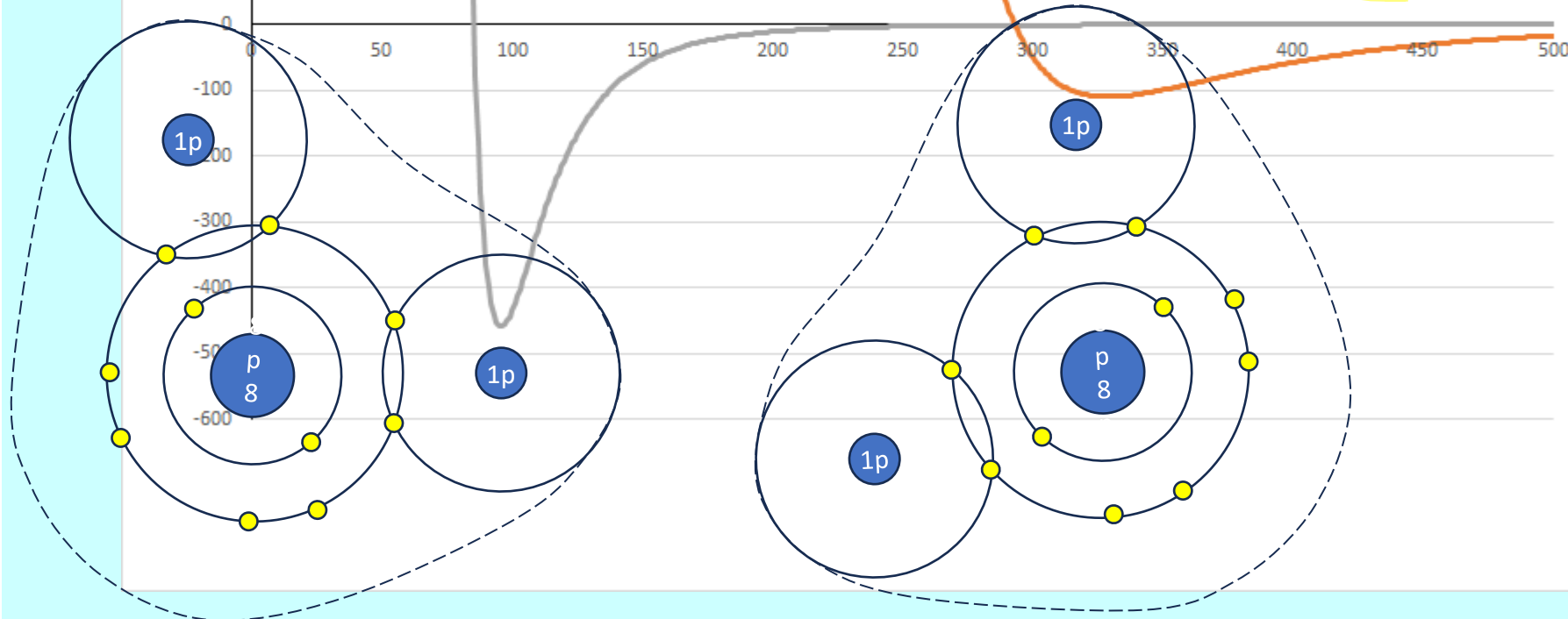
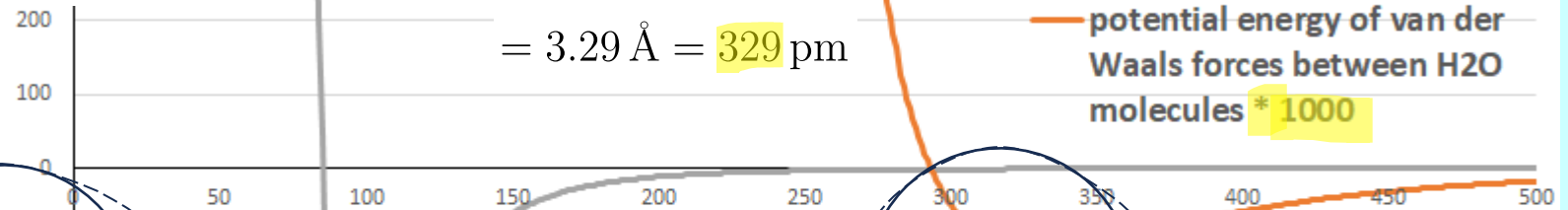
$V(r)$  in kJ/mol vs  $r$  in pm ( $10^{-12} \text{ m}$ )

$$\epsilon = 0.112 \text{ kJ/mol}$$

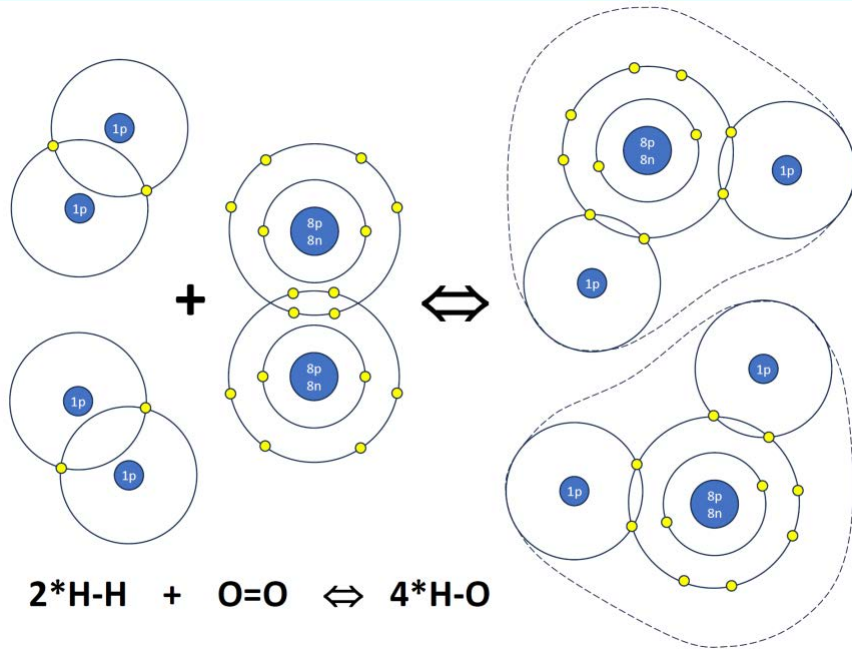
$$r_{\min} = 2^{1/6} \sigma = 3.29 \text{ \AA} = 329 \text{ pm}$$

— potential energy for an H-O bond in a water molecule

— potential energy of van der Waals forces between H<sub>2</sub>O molecules \* 1000



# Enthalpy of formation and bond energies



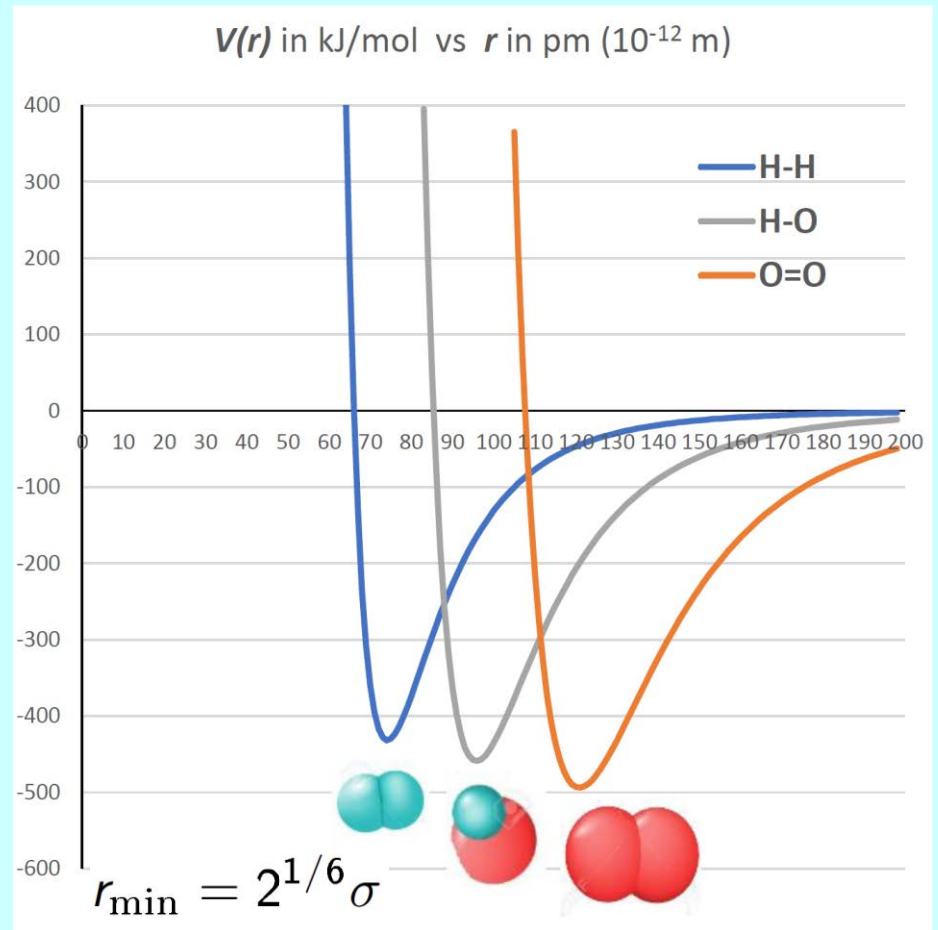
Bond parameters	Lennard Jones		H-H	O=O	H-O
Strength	$\epsilon$	kJ/mol	<b>432</b>	<b>494</b>	<b>459</b>
Length	$r_{\min}$	$10^{-12}$ m	<b>74.1</b>	<b>120.8</b>	<b>95.8</b>

$$2*\epsilon(\text{H-H}) + \epsilon(\text{O=O}) - 4*\epsilon(\text{H-O}) \quad \mathbf{-478}$$

Enthalpy of formation of 2 moles of water **-484**

$$V_{\text{LJ}}(r) = 4\epsilon \left[ \left( \frac{\sigma}{r} \right)^{12} - \left( \frac{\sigma}{r} \right)^6 \right]$$

$$= \epsilon \left[ \left( \frac{r_{\min}}{r} \right)^{12} - 2 \left( \frac{r_{\min}}{r} \right)^6 \right]$$





# Properties of formation and Hess relation

Define a complete set of independent elemental constituents  $A_j^e$ :

**Completeness:** for every molecule  $A_i$  there must be a reaction of formation from the  $A_j^e$ 's:

$$\sum_j \alpha_{ij} A_j^e = A_i \quad \text{or in our notation} \quad - \sum_j \alpha_{ij} A_j^e + A_i = 0 \quad \text{or also} \quad A_i = \sum_j \alpha_{ij} A_j^e$$

**Independence:** the  $A_j^e$ 's cannot be transformed into one another:

$$\sum_j \beta_j A_j^e = 0 \quad \Rightarrow \quad \beta_j = 0 \quad \forall j$$

**Practical choice:**  $\{A_j^e\}$  = the species with nuclei of only one type that are most stable at STP, one for each atomic number  $Z_j^e$  in the periodic table (isotopes are chemically equivalent).

**Property of formation at STP** of a generic chemical species  $A_i$ :

$$\Delta u_f^o|_i = - \sum_j \alpha_{ij} u_{jj}^e(T_o, p_o) + u_{ii}(T_o, p_o)$$

We can safely select  $u_{jj}^e(T_o, p_o) = 0$ : in fact, for any mechanism  $\sum_i \nu_i A_i = 0$  we can write:

$$\sum_i \nu_i A_i = \sum_i \nu_i \sum_j \alpha_{ij} A_j^e = \sum_j \left( \sum_i \nu_i \alpha_{ij} \right) A_j^e = 0 \quad \Rightarrow \quad \sum_i \nu_i \alpha_{ij} = 0 \quad \forall j$$

$$\Delta u^o = \sum_i \nu_i u_{ii}(T_o, p_o) = \sum_i \nu_i \left[ \Delta u_f^o|_i + \sum_j \alpha_{ij} u_{jj}^e(T_o, p_o) \right] = \sum_i \nu_i \Delta u_f^o|_i + \sum_j \overbrace{\sum_i \nu_i \alpha_{ij}} u_{jj}^e(T_o, p_o)$$

Therefore, we conventionally choose these references:

$$\begin{aligned} u_{jj}^e(T_o, p_o) &= 0 && \text{for the } A_j^e \text{'s and} \\ u_{ii}(T_o, p_o) &= \Delta u_f^o|_i && \text{for all other species} \end{aligned}$$

and use the **Hess relation**

$$\Delta u^o = \sum_i \nu_i \Delta u_f^o|_i$$

Careful! Properties of formation at STP are listed for the **fictitious ideal gas state** by extrapolating the ideal-gas correlations for  $c_{p,ii}(T)$  from the  $T, p$  range where they hold, down to  $T_o, p_o$ :

$$h_{ii}(T_o, p_o) = h_{ii}(T, p_o) - \int_{T_o}^T c_{p,ii}(T') dT' \quad s_{ii}(T_o, p_o) = s_{ii}(T, p_o) - \int_{T_o}^T \frac{c_{p,ii}(T')}{T'} dT' - R \ln \frac{p_o}{p}$$

# Properties of formation and Hess relation

Hess relation  $\Delta h^o = \sum_i \nu_i \Delta h_f^o|_i$

For the reaction  $\text{CH}_4 + 2 \text{O}_2 = \text{CO}_2 + 2 \text{H}_2\text{O}$

$M_{\text{CH}_4} = 16 \text{ kg/kmol}$

$$-1(-74.9) - 2(0) + 1(-393.8) + 2(-242) = -802.9 \text{ MJ/kmol}_{\text{CH}_4} = -50.2 \text{ MJ/kg}_{\text{CH}_4}$$

For the reaction  $\text{C}_8\text{H}_{18} + 12.5 \text{O}_2 = 8 \text{CO}_2 + 9 \text{H}_2\text{O}$

$M_{\text{C}_8\text{H}_{18}} = 114 \text{ kg/kmol}$

$$-1(-224.3) - 12.5(0) + 8(-393.8) + 9(-242) = -5120 \text{ MJ/kmol}_{\text{C}_8\text{H}_{18}} = -44.9 \text{ MJ/kg}_{\text{C}_8\text{H}_{18}}$$

Substance	Formula	$\Delta h_f^o$ MJ/kmol	$\Delta g_f^o$ MJ/kmol	$\Delta s_f^o$ kJ/kmol K
Acetic acid	$\text{C}_2\text{H}_4\text{O}_2$	-336.5	-286.3	-168.4
Acetone	$\text{CH}_3\text{COCH}_3$	-217.7	-153.2	-216.5
Acetylene	$\text{C}_2\text{H}_2$	226.9	209.3	58.8
Ammonia	$\text{NH}_3$	-45.7	-16.2	-99.1
Argon	Ar	0	0	0
Benzene	$\text{C}_6\text{H}_6$	83.0	129.7	-156.9
Carbon	C	0	0	0
Carbon dioxide	$\text{CO}_2$	-393.8	-394.6	2.9
Carbon monoxide	CO	-110.6	-137.4	89.7
Chlorine	$\text{Cl}_2$	0	0	0
Chloroform	$\text{CHCl}_3$	-101.3	-68.6	-109.8
Ethane	$\text{C}_2\text{H}_6$	-84.7	-33.0	-173.7
Ethanol	$\text{C}_2\text{H}_5\text{OH}$	-235.0	-168.4	-223.3
Ethylene	$\text{C}_2\text{H}_4$	52.3	68.2	-53.1
Fluorine	$\text{F}_2$	0	0	0
Freon 12	$\text{CCl}_2\text{F}_2$	-481.5	-442.5	-130.6
Freon 13	$\text{CClF}_3$	-695.0	-654.4	-136.2
Freon 21	$\text{CHCl}_2\text{F}$	-298.9	-268.4	-102.5
Freon 22	$\text{CHClF}_2$	-502.0	-470.9	-104.3

Substance	Formula	$\Delta h_f^o$ MJ/kmol	$\Delta g_f^o$ MJ/kmol	$\Delta s_f^o$ kJ/kmol K
Hydrogen	$\text{H}_2$	0	0	0
Hydrogen (atomic)	H	218.0	203.3	49.4
Hydrogen chloride	HCl	-92.4	-95.3	10.0
Hydroxyl	OH	39.5	34.3	17.4
Isooctane	$\text{C}_8\text{H}_{18}$	-224.3	13.7	-798.6
Methane	$\text{CH}_4$	-74.9	-50.9	-80.6
Methanol	$\text{CH}_3\text{OH}$	-201.3	-162.6	-129.8
Methylene chloride	$\text{CH}_2\text{Cl}_2$	-95.5	-68.9	-89.0
Naphthalene	$\text{C}_{10}\text{H}_8$	151.1	223.7	-243.8
Nitric oxide	NO	90.4	86.8	12.4
Nitrogen	$\text{N}_2$	0	0	0
Nitrogen (atomic)	N	472.8	455.6	57.6
Nitrogen dioxide	$\text{NO}_2$	33.9	52.0	-60.8
Nitrous oxide	$\text{N}_2\text{O}$	81.6	103.7	-74.1
n-Octane	$\text{C}_8\text{H}_{18}$	-208.6	16.4	-754.6
Oxygen	$\text{O}_2$	0	0	0
Oxygen (atomic)	O	249.2	231.8	58.3
Ozone	$\text{O}_3$	142.8	162.9	-67.5
Propane	$\text{C}_3\text{H}_8$	-103.9	-23.5	-269.8
Propylene	$\text{CH}_2\text{CHCH}_3$	20.4	62.8	-142.0
Water	$\text{H}_2\text{O}$	-242.0	-228.8	-44.4

Adapted from Table 29.4 of Gyftopoulos, Beretta, Thermodynamics. Foundations and applications, Dover, 2005

# **Systems with chemical reactions**

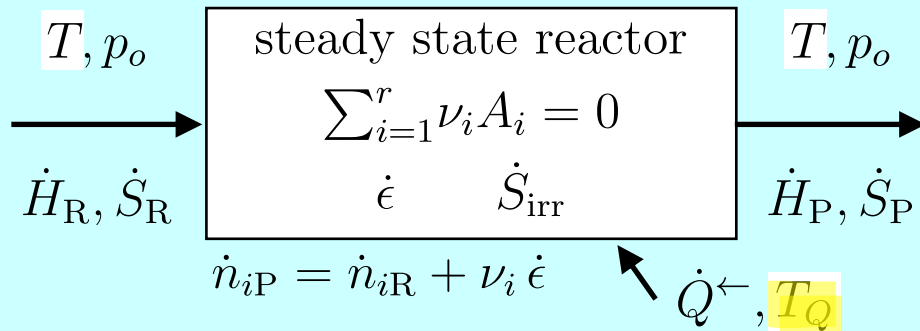
**what determines the direction  
of spontaneous reaction  
and  
how temperature affects it**

**Exergy**

**Maximum work obtainable**

**Minimum work required**

# When is an isothermobaric reaction product- or reactant-favored?



Assume:

- steady-state
- heat interaction at  $T_Q$  maintains inlet and outlet temperatures at  $T$
- in and out streams are either pure component or ideal GD mixtures
- neglect mixing terms in the entropies

$$0 = \dot{H}_R(T) - \dot{H}_P(T) + \dot{Q}^{\leftarrow}$$

$$0 = \dot{S}_R(T) - \dot{S}_P(T) + \frac{\dot{Q}^{\leftarrow}}{T_Q} + \dot{S}_{\text{irr}}$$

$$T_Q \dot{S}_{\text{irr}} = -\dot{\epsilon} \underbrace{[\Delta h^{\circ}(T) - T_Q \Delta s^{\circ}(T)]}_{\text{this must be negative for the reaction to be product-favored}} \geq 0$$

this must be negative  
for the reaction to be  
product-favored

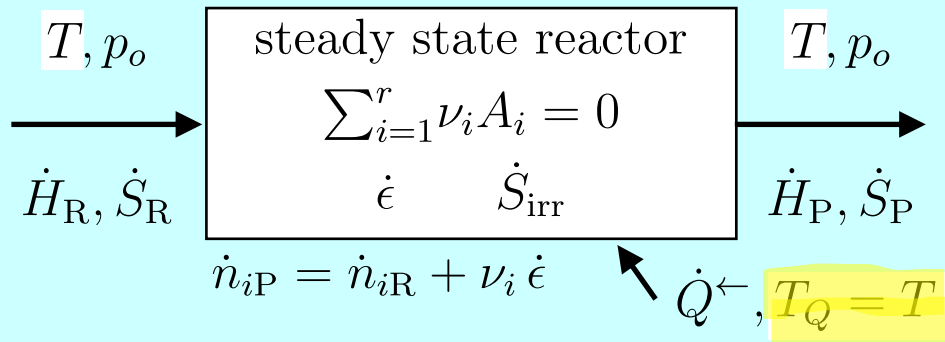
$$\dot{H}_P(T) - \dot{H}_R(T) = \dot{\epsilon} \Delta h^{\circ}(T)$$

$$\dot{S}_P(T) - \dot{S}_R(T) = \dot{\epsilon} \Delta s^{\circ}(T)$$

$$\frac{\dot{S}_{\text{irr}}}{R \dot{\epsilon}} = -\frac{\Delta h^{\circ}(T) - T_Q \Delta s^{\circ}(T)}{R T_Q}$$

$\Delta h^{\circ}(T)$	$\Delta s^{\circ}(T)$	$\dot{\epsilon}$
$< 0$ (exothermic)	$> 0$	$> 0$ (at any temperature)
$> 0$ (endothermic)	$< 0$	$< 0$ (at any temperature)
$< 0$ (exothermic)	$< 0$	$> 0$ (if $-\Delta s^{\circ}(T) < -\Delta h^{\circ}(T)/T_Q$ )
$> 0$ (endothermic)	$> 0$	$< 0$ (if $\Delta s^{\circ}(T) < \Delta h^{\circ}(T)/T_Q$ )

# When is an isothermobaric reaction product- or reactant-favored?



Assume:

- steady-state
- heat interaction at  $T_Q$  maintains inlet and outlet temperatures at  $T$
- in and out streams are either pure component or ideal GD mixtures
- neglect mixing terms in the entropies

$$0 = \dot{H}_R(T) - \dot{H}_P(T) + \dot{Q}^{\leftarrow}$$

$$0 = \dot{S}_R(T) - \dot{S}_P(T) + \frac{\dot{Q}^{\leftarrow}}{T} + \dot{S}_{\text{irr}}$$

$$T_Q \dot{S}_{\text{irr}} = -\dot{\epsilon} [\Delta h^\circ(T) - T \Delta s^\circ(T)] = -\dot{\epsilon} \Delta g^\circ(T) \geq 0$$

$\Delta g^\circ(T)$  must be negative  
for the reaction to be  
product-favored

$$\dot{H}_P(T) - \dot{H}_R(T) = \dot{\epsilon} \Delta h^\circ(T)$$

$$\dot{S}_P(T) - \dot{S}_R(T) = \dot{\epsilon} \Delta s^\circ(T)$$

$$\frac{\dot{S}_{\text{irr}}}{R \dot{\epsilon}} = -\frac{\Delta g^\circ(T)}{RT}$$

$$\frac{d}{dT} \frac{\dot{S}_{\text{irr}}}{R \dot{\epsilon}} = -\frac{d}{dT} \frac{\Delta g^\circ(T)}{RT} = \frac{\Delta h^\circ(T)}{RT^2}$$

$\Delta h^\circ(T)$	$\Delta g^\circ(T)$	$\dot{\epsilon}$
$< 0$ (exothermic)	$< 0$	$> 0$ (but can go negative for lower $T$ )
$> 0$ (endothermic)	$< 0$	$> 0$ (but can go negative for higher $T$ )
$< 0$ (exothermic)	$> 0$	$< 0$ (but can go positive for higher $T$ )
$> 0$ (endothermic)	$> 0$	$< 0$ (but can go positive for lower $T$ )

# When is an isothermal reaction product- or reactant-favored?

$$\begin{aligned} \text{Recall } g_{ii} &= \mu_{ii} \text{ and } h_{ii} = \left( \frac{\partial(\mu_{ii}/T)}{\partial(1/T)} \right)_p \\ \frac{d}{d(1/T)} \frac{\Delta g^\circ(T)}{RT} &= \frac{d}{d(1/T)} \frac{\sum_i \nu_i \mu_{ii}(T, p_o)}{RT} \\ &= \sum_i \nu_i \frac{d}{d(1/T)} \frac{\mu_{ii}(T, p_o)}{RT} \\ &= \sum_i \nu_i \frac{h_{ii}(T, p_o)}{R} = \frac{\Delta h^\circ(T)}{R} \end{aligned}$$

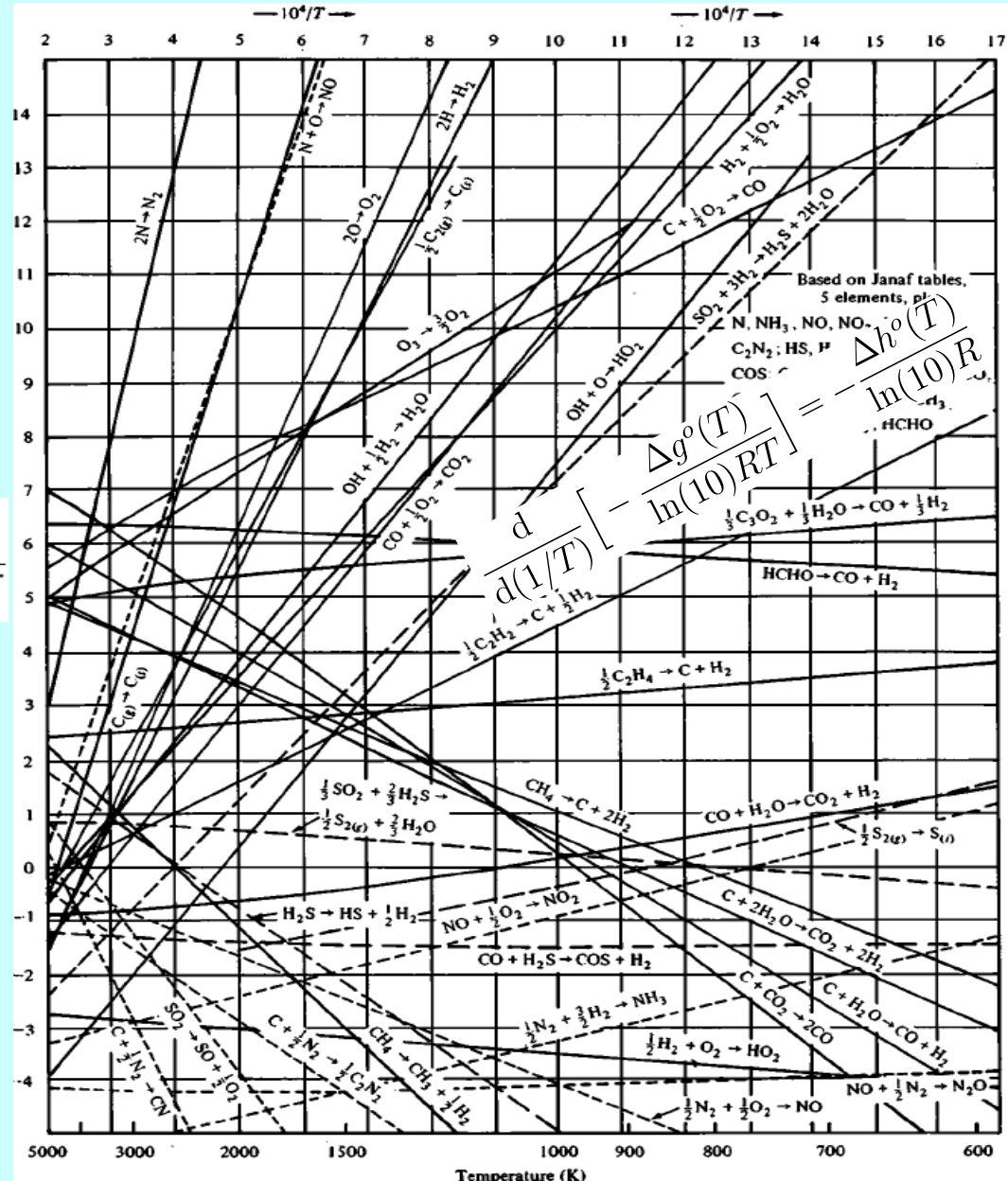
$$\frac{\dot{S}_{\text{irr}}}{\ln(10)R\dot{\epsilon}} = - \frac{\Delta g^\circ(T)}{\ln(10)RT}$$

product-favored  $\uparrow$

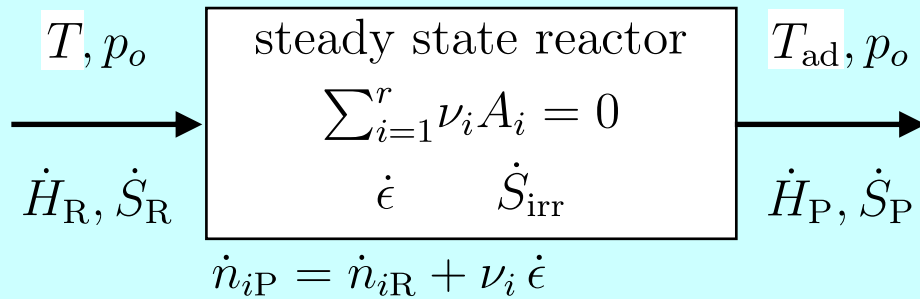
For  $T = T_Q$ ,  $\longleftarrow$

reactant-favored  $\downarrow$

Adapted from Figure 30.1 of Gyftopoulos, Beretta, Thermodynamics. Foundations and applications, Dover, 2005



# When is an adiabatic reaction product- or reactant-favored?



Assume:

- steady-state
- adiabatic, no heat interactions
- in and out streams are either pure component or ideal GD mixtures
- neglect mixing terms in the entropies

$$0 = \dot{H}_R(T) - \dot{H}_P(T_{\text{ad}}) \Rightarrow \text{solve for } T_{\text{ad}}$$

$$0 = \dot{S}_R(T) - \dot{S}_P(T_{\text{ad}}) + \dot{S}_{\text{irr}}$$

$$T \dot{S}_{\text{irr}} = -\dot{\epsilon} \Delta g^o(T) - \dot{E}_{\text{XP}}(T_{\text{ad}}) \geq 0$$

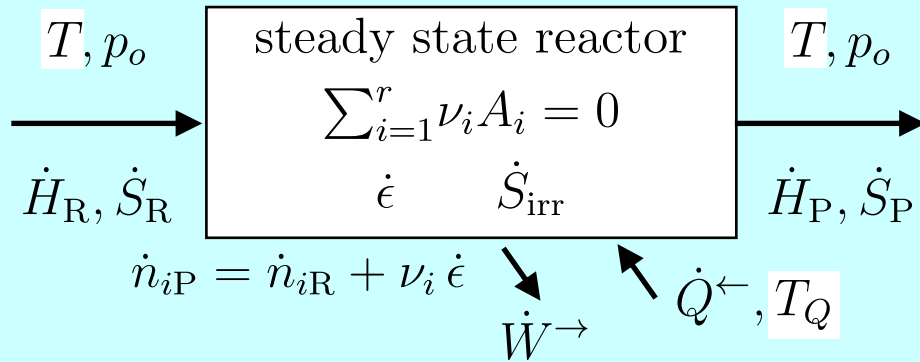
For  $\dot{\epsilon} > 0$ ,  $\Delta g^o(T)$  must be negative enough that  $-\dot{\epsilon} \Delta g^o(T)$  is greater than the exergy\* needed to heat/cool the products from environmental  $T$  to  $T_{\text{ad}}$

$$\dot{H}_P(T) - \dot{H}_R(T) = \dot{\epsilon} \Delta h^o(T)$$

$$\dot{S}_P(T) - \dot{S}_R(T) = \dot{\epsilon} \Delta s^o(T)$$

$$\dot{E}_{\text{XP}}(T_{\text{ad}}) = \dot{H}_P(T_{\text{ad}}) - \dot{H}_P(T) - T [\dot{S}_P(T_{\text{ad}}) - \dot{S}_P(T)]$$

# Exergy / Maximum work obtainable / Minimum work required



Assume:

- steady-state, with a work interaction
- heat interaction at  $T_Q$  maintains inlet and outlet temperatures at  $T$
- in and out streams are either pure component or ideal GD mixtures
- neglect mixing terms in the entropies

$$0 = \dot{H}_R(T) - \dot{H}_P(T) + \dot{Q}^{\leftarrow} - \dot{W}^{\rightarrow}$$

$$0 = \dot{S}_R(T) - \dot{S}_P(T) + \frac{\dot{Q}^{\leftarrow}}{T_Q} + \dot{S}_{\text{irr}}$$

$$\dot{W}^{\rightarrow} = -\dot{\epsilon} \underbrace{[\Delta h^{\circ}(T) - T_Q \Delta s^{\circ}(T)]}_{\text{this must be negative for } \dot{\epsilon} > 0 \text{ and } \dot{W}^{\rightarrow} > 0} - T_Q \dot{S}_{\text{irr}}$$

this must be negative  
for  $\dot{\epsilon} > 0$  and  $\dot{W}^{\rightarrow} > 0$

Equivalently,

$$\dot{W}^{\leftarrow} = \dot{\epsilon} \underbrace{[\Delta h^{\circ}(T) - T_Q \Delta s^{\circ}(T)]}_{\text{if this is positive } \dot{\epsilon} > 0 \text{ requires } \dot{W}^{\leftarrow} > 0} + T_Q \dot{S}_{\text{irr}}$$

if this is positive  
 $\dot{\epsilon} > 0$  requires  $\dot{W}^{\leftarrow} > 0$

$$\dot{H}_P(T) - \dot{H}_R(T) = \dot{\epsilon} \Delta h^{\circ}(T)$$

$$\dot{S}_P(T) - \dot{S}_R(T) = \dot{\epsilon} \Delta s^{\circ}(T)$$

$$\dot{W}_{\text{max}}^{\rightarrow} = -\dot{\epsilon} [\Delta h^{\circ}(T) - T_Q \Delta s^{\circ}(T)]$$

$$\dot{W}_{\text{min}}^{\leftarrow} = \dot{\epsilon} [\Delta h^{\circ}(T) - T_Q \Delta s^{\circ}(T)]$$



# **Systems with chemical reactions**

**Oxidation of hydrocarbons**

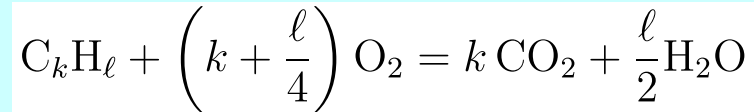
**Heating values of fuels**

**Exergy of a fuel**

**Adiabatic flame temperature**

# Oxidation of hydrocarbons

Overall basic reaction mechanism, assuming full oxidation



If oxygen is taken from dry air with mole fraction  $y_{\text{O}_2} = 0.2095$ , 1 mol of  $\text{O}_2$  requires 4.77 mol of air. From this, the **stoichiometric air-fuel ratio** is defined as

$$\left(\frac{n_{\text{air}}}{n_{\text{fuel}}}\right)_s = 4.77 \left(k + \frac{\ell}{4}\right) \quad \left(\frac{m_{\text{air}}}{m_{\text{fuel}}}\right)_s = 4.77 \left(k + \frac{\ell}{4}\right) \frac{28.96}{M_{\text{fuel}}}$$

Moreover, in order to increase the probability of having the oxidation of all the molecules, an excess of air is usually used i.e.  $n_{\text{air}} = (n_{\text{air}})_s + (n_{\text{air}})_{\text{excess}}$ , so that the **equivalence ratio** is defined

$$\lambda = \frac{n_{\text{air}}/n_{\text{fuel}}}{(n_{\text{air}}/n_{\text{fuel}})_s} = \frac{m_{\text{air}}/m_{\text{fuel}}}{(m_{\text{air}}/m_{\text{fuel}})_s} > 1$$

For most hydrocarbons (except  $\text{H}_2$ )

$$-2.4\% < \frac{\Delta h^0 - \Delta g^0}{\Delta g^0} < +2.4\%$$

and

$$40 \text{ MJ/kg} < \text{LHV} = -\Delta h^0 < 50 \text{ MJ/kg}$$

# Properties of the reaction of oxidation of hydrocarbons

Fuel	Formula	$M$	$\Delta h^\circ$	$\Delta g^\circ$	$\Delta s^\circ$	$\frac{\Delta h^\circ - \Delta g^\circ}{\Delta g^\circ}$
		$\frac{\text{kg}}{\text{kmol}}$	$\frac{\text{MJ}}{\text{kg}}$	$\frac{\text{MJ}}{\text{kg}}$	$\frac{\text{kJ}}{\text{kg K}}$	%
Hydrogen	H <sub>2</sub>	2.016	-120.0	-113.5	-22.0	+5.8
Carbon (graphite)	C	12.011	-32.8	-32.9	0.2	-0.2
Methane	CH <sub>4</sub>	16.043	-50.0	-49.9	-0.3	+0.2
Acetylene	C <sub>2</sub> H <sub>2</sub>	26.038	-48.3	-47.1	-3.7	+2.4
Ethylene	C <sub>2</sub> H <sub>4</sub>	28.054	-47.2	-46.9	-1.1	+0.7
Ethane	C <sub>2</sub> H <sub>6</sub>	30.07	-47.5	-48.0	1.5	-1.0
Propylene	C <sub>3</sub> H <sub>6</sub>	42.081	-45.8	-45.9	0.4	-0.3
Propane	C <sub>3</sub> H <sub>8</sub>	44.097	-46.4	-47.1	2.3	-1.5
<i>n</i> -Butane	C <sub>4</sub> H <sub>10</sub>	58.12	-45.8	-46.6	2.7	-1.7
<i>n</i> -Pentane	C <sub>5</sub> H <sub>12</sub>	72.15	-45.4	-46.3	2.9	-1.9
Benzene	C <sub>6</sub> H <sub>6</sub>	78.114	-40.6	-40.8	0.5	-0.4
<i>n</i> -Hexane	C <sub>6</sub> H <sub>14</sub>	86.18	-45.1	-46.1	3.1	-2.0
<i>n</i> -Heptane	C <sub>7</sub> H <sub>16</sub>	100.21	-45.0	-45.9	3.2	-2.1
<i>n</i> -Octane	C <sub>8</sub> H <sub>18</sub>	114.232	-44.8	-45.8	3.3	-2.2
Isooctane	C <sub>8</sub> H <sub>18</sub>	114.232	-44.7	-45.8	3.7	-2.4
<i>n</i> -Nonane	C <sub>9</sub> H <sub>20</sub>	128.26	-44.7	-45.7	3.4	-2.2
<i>n</i> -Decane	C <sub>10</sub> H <sub>22</sub>	142.29	-44.6	-45.7	3.5	-2.3
Carbon monoxide	CO	28.01	-10.1	-9.2	-3.1	+10.1
Methanol	CH <sub>3</sub> OH	32.042	-21.1	-21.5	1.4	-1.9
Ethanol	C <sub>2</sub> H <sub>5</sub> OH	46.069	-27.8	-28.4	2.1	-2.2
Ethylene glycol	(CH <sub>2</sub> OH) <sub>2</sub>	62.07	-17.1	-18.6	5.1	-8.1

Adapted from Figure 31.7 of Gyftopoulos, Beretta, Thermodynamics. Foundations and applications, Dover, 2005

# Heating values of fuels

Fuel	Formula	$M$ $\frac{\text{kg}}{\text{kmol}}$	$\rho$ $\frac{\text{kg}}{\text{liter}}$	$h_{fg}$ $\frac{\text{kJ}}{\text{kg}}$	$(c_p)_f$ $\frac{\text{kJ}}{\text{kg K}}$	$(c_p)_g$ $\frac{\text{kJ}}{\text{kg K}}$	HHV $\frac{\text{MJ}}{\text{kg}}$	LHV $\frac{\text{MJ}}{\text{kg}}$	$(A/F)_s$
<b>Practical fuels</b> (Typical composition)									
Gasoline	$C_{7.92}H_{14.8}$ (l)	110	0.75	305	2.4	1.7	47.3	44.0	14.6
Light diesel	$C_{12.3}H_{22.1}$ (l)	170	0.86	270	2.2	1.7	44.8	42.5	14.5
Heavy diesel	$C_{14.6}H_{24.8}$ (l)	200	0.89	230	1.9	1.7	43.8	41.4	14.4
Natural gas	$C_{1.1}H_{3.9}N_{0.1}$ (g)	18	—	—	—	2	50.0	45.0	14.5
<b>Hydrocarbons</b>									
Methane	$CH_4$ (g)	16.04	—	—	—	2.2	55.5	50.0	17.23
Propane	$C_3H_8$ (g)	44.10	—	—	—	1.6	50.4	46.4	15.67
Isooctane	$C_8H_{18}$ (l)	114.23	0.692	308	2.1	1.6	47.8	44.3	15.13
Cetane	$C_{16}H_{34}$ (l)	226.44	0.773	358	—	1.6	47.3	44.0	14.82
Benzene	$C_6H_6$ (l)	78.11	0.879	433	1.7	1.1	41.9	40.2	13.27
Toluene	$C_7H_8$ (l)	92.14	0.867	412	1.7	1.1	42.5	40.6	13.50
<b>Alcohols</b>									
Methanol	$CH_3OH$ (l)	32.04	0.792	1103	2.6	1.7	22.7	20.0	6.47
Ethanol	$C_2H_5OH$ (l)	46.07	0.785	840	2.5	1.9	29.7	26.9	9.00
<b>Other fuels</b>									
Carbon	$C$ (s)	12.01	2	—	—	—	33.8	33.8	11.51
Carbon monoxide	$CO$ (g)	28.01	—	—	—	1.1	10.1	10.1	2.467
Hydrogen	$H_2$ (g)	2.015	—	—	—	1.4	142.0	120.0	34.3

Source: Data mainly from J. B. Heywood, *Internal Combustion Engine Fundamentals*, McGraw-Hill, New York, 1988.

<sup>a</sup> Values of  $h_{fg}$  at 25°C for liquid fuels. Values of  $\rho$  and  $(c_p)_f$  at 25°C and 1 atm for liquid fuels. Values of  $(c_p)_g$  at 25°C and saturation pressure for liquid fuels, and at 1 atm and 25°C for gaseous fuels.

Adapted from Figure 31.1 of Gyftopoulos, Beretta, *Thermodynamics. Foundations and applications*, Dover, 2005

# Lower and higher heating values of fuels

In combustion, water is present both in reactants (moist air) and in products, this complicates the stoichiometry. The highest possible mole fraction of water vapor in the gaseous mixture of products of combustion is given by Raoult's law ( $1 = \text{H}_2\text{O}$ )

$$y_{1g}^{\max} p = p_{\text{sat},11}^{\text{fg}}(T) \text{ where } y_{1f} = 1 \text{ (condensate is pure liquid water)}$$

Let  $\phi = y_{1g}/y_{1g}^{\max}$  be the relative humidity of the products of combustion in the gaseous phase. Since  $p_{\text{sat},11}^{\text{fg}}(25^\circ\text{C}) = 3.169 \text{ kPa}$ , at STP ( $T_o = 25^\circ\text{C}$  and  $p_o = 1 \text{ atm}$ ), the maximum amount of vapor is  $y_{1g}^{\max} = 3.169/101.325 = 0.0316$ , so that by splitting water vapor and condensate, i.e.,  $n_{\text{H}_2\text{O}} = n_{\text{H}_2\text{O},g} + n_{\text{H}_2\text{O},f}$  we get

$$\left. \begin{aligned} \frac{n_{\text{H}_2\text{O}}}{n_{\text{C}_k\text{H}_\ell}} &= \frac{n_{\text{H}_2\text{O},f}}{n_{\text{C}_k\text{H}_\ell}} + \frac{n_{\text{H}_2\text{O},g}}{n_{\text{C}_k\text{H}_\ell}} = \frac{\ell}{2} \\ \frac{n_{\text{tot,prod}}}{n_{\text{C}_k\text{H}_\ell}} &= \frac{\ell}{4} + 4.77 \left( k + \frac{\ell}{4} \right) \\ \frac{n_{\text{H}_2\text{O},g}}{n_{\text{tot,prod}} - n_{\text{H}_2\text{O},f}} &= y_{1g} \leq 0.0316 \end{aligned} \right\} \Rightarrow \begin{aligned} \frac{n_{\text{H}_2\text{O},g}}{n_{\text{C}_k\text{H}_\ell}} &\leq \frac{0.0316 \left[ 4.77 \left( k + \frac{\ell}{4} \right) - \frac{\ell}{4} \right]}{1 - 0.0316} = 0.123 \ell + 0.156 k \\ \text{so that when } \ell/2 &\text{ exceeds this amount, i.e., when} \\ \ell/k &> 0.414, \text{ then part of the water condenses} \\ \frac{n_{\text{H}_2\text{O},f}}{n_{\text{C}_k\text{H}_\ell}} &= \frac{\frac{\ell}{2} - 0.0316 \left[ \frac{\ell}{2} + 4.77 \left( k + \frac{\ell}{4} \right) \right]}{1 - 0.0316} = 0.469 \ell - 0.156 k \end{aligned}$$

meaning that the  $\dot{Q}_o^\rightarrow$  extracted is not only given by  $\dot{n}_{\text{C}_k\text{H}_\ell} \Delta h^\circ$  but also by the condensation enthalpy

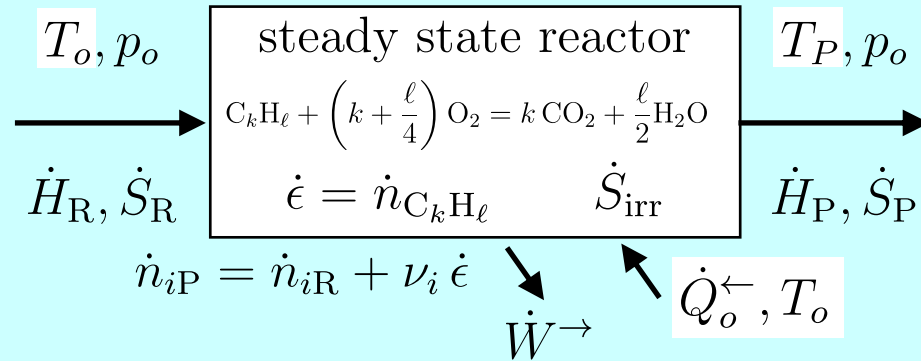
$$\dot{Q}_o^\rightarrow = \dot{H}_R^\circ - \dot{H}_P^\circ = \dot{n}_{\text{C}_k\text{H}_\ell} \text{HV}_{\text{actual}} \quad \text{where} \quad \text{HV}_{\text{actual}} = -\Delta h^\circ + \frac{\dot{n}_{\text{H}_2\text{O},f}}{\dot{n}_{\text{C}_k\text{H}_\ell}} h_{\text{H}_2\text{O},\text{fg}}(T_o)$$

With less details, in the common practice one defines the lower and higher heating values

$$\text{LHV} = -\Delta h^\circ \quad \text{and} \quad \text{HHV} = -\Delta h^\circ + \frac{\ell}{2} h_{\text{H}_2\text{O},\text{fg}}(T_o)$$

For example, for methane, LHV,  $\text{HV}_{\text{actual}}$ , and HHV are, respectively, 50.1, 54.7, and 55.5 MJ/kg.

# Heating value and Exergy of a fuel / Adiabatic flame temperature



**Heating value** setup:  $T_P = T_o$ ,  $\dot{W}^{\rightarrow} = 0$ ,  $\dot{e} = \dot{n}_{C_k H_l}$

$$\dot{Q}_o^{\rightarrow} = \dot{H}_R^o - \dot{H}_P^o = \dot{e} HV_{\text{actual}} \quad \text{where} \quad HV_{\text{actual}} = -\Delta h^o + \frac{\dot{n}_{\text{H}_2\text{O},f}}{\dot{n}_{C_k H_l}} h_{\text{H}_2\text{O},fg}(T_o)$$

**Exergy** setup:  $T_P = T_o$ ,  $\dot{W}^{\rightarrow} = \dot{W}_{\text{max}}$ ,  $\dot{S}_{\text{irr}} = 0$

$$\dot{W}_{\text{max}} = \dot{H}_R^o - \dot{H}_P^o - T_o (\dot{S}_R^o - \dot{S}_P^o) = \dot{e} EX_{C_k H_l} \quad \text{where} \quad EX_{C_k H_l} = -\Delta h^o + T_o \Delta s^o = -\Delta g^o$$

Notice that condensation does not contribute because  $h_{\text{H}_2\text{O},fg}(T_o) - T_o s_{\text{H}_2\text{O},fg}(T_o) = 0$ .

**Adiabatic flame temperature** setup:  $T_P = T_{\text{ad}}$ ,  $\dot{Q}_o^{\rightarrow} = 0$ ,  $\dot{W}^{\rightarrow} = 0$ , assume ideal GD mix of ideal gases

$$0 = \dot{H}_R^o - \dot{H}_P(T_{\text{ad}}) = \dot{H}_R^o - \dot{H}_P^o + \dot{H}_P^o - \dot{H}_P(T_{\text{ad}}) = \dot{e} HV_{\text{actual}} - \dot{n}_P \int_{T_o}^{T_{\text{ad}}} \sum_i y_{iP} c_{p,ii}(T') dT' \Rightarrow \text{solve for } T_{\text{ad}}$$

$$\dot{S}_{\text{irr}} = \dot{S}_P(T_{\text{ad}}) - \dot{S}_R^o = \dot{S}_P(T_{\text{ad}}) - \dot{S}_P^o + \dot{S}_P^o - \dot{S}_R^o = \dot{n}_P \int_{T_o}^{T_{\text{ad}}} \frac{\sum_i y_{iP} c_{p,ii}(T')}{T'} dT' + \dot{e} \Delta s^o \approx -0.3 \dot{e} \frac{\Delta g^o}{T_o}$$

Typical values of  $\frac{T_o \dot{S}_{\text{irr}}}{-\dot{e} \Delta g^o}$  are in the range 26% to 30%

# Adiabatic flame temperature calculations

$$c_{p,ii}(T) = a_{ii} + b_{ii}T^{1/4} + c_{ii}T^{1/2} + d_{ii}T^{3/4}$$

$$\sum_i y_{iP} c_{p,ii}(T) = a + bT^{1/4} + cT^{1/2} + dT^{3/4}$$

$$\text{where } a = \sum_i y_{iP} a_{ii} \quad b = \sum_i y_{iP} b_{ii} \quad c = \sum_i y_{iP} c_{ii} \quad d = \sum_i y_{iP} d_{ii}$$

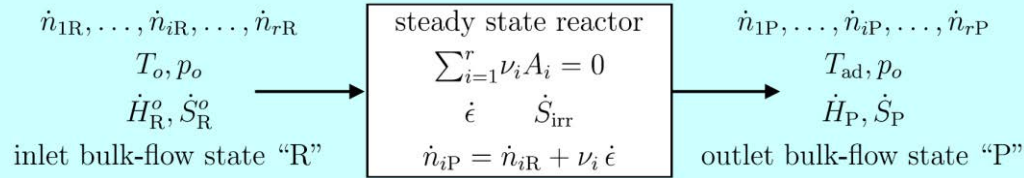
$$\int_{T_o}^{T_{ad}} \sum_i y_{iP} c_{p,ii}(T') dT' = a(T_{ad} - T_o) + \frac{4}{5}b(T_{ad}^{5/4} - T_o^{5/4}) + \frac{2}{3}c(T_{ad}^{3/2} - T_o^{3/2}) + \frac{4}{7}d(T_{ad}^{7/4} - T_o^{7/4})$$

$$\int_{T_o}^{T_{ad}} \frac{\sum_i y_{iP} c_{p,ii}(T')}{T'} dT' = a \ln \frac{T_{ad}}{T_o} + 4b(T_{ad}^{1/4} - T_o^{1/4}) + 2c(T_{ad}^{1/2} - T_o^{1/2}) + \frac{4}{3}d(T_{ad}^{3/4} - T_o^{3/4})$$

Substance	Formula	<i>a</i>	<i>b</i>	<i>c</i>	<i>d</i>
Carbon dioxide	CO <sub>2</sub>	-55.6	30.5	-1.96	0
Carbon monoxide	CO	62.8	-22.6	4.6	-0.272
Oxygen	O <sub>2</sub>	10.3	5.4	-0.18	0
Nitrogen	N <sub>2</sub>	72	-26.9	5.19	-0.298
Hydrogen	H <sub>2</sub>	79.5	-26.3	4.23	-0.197
Water	H <sub>2</sub> O	180	-85.4	15.6	-0.858

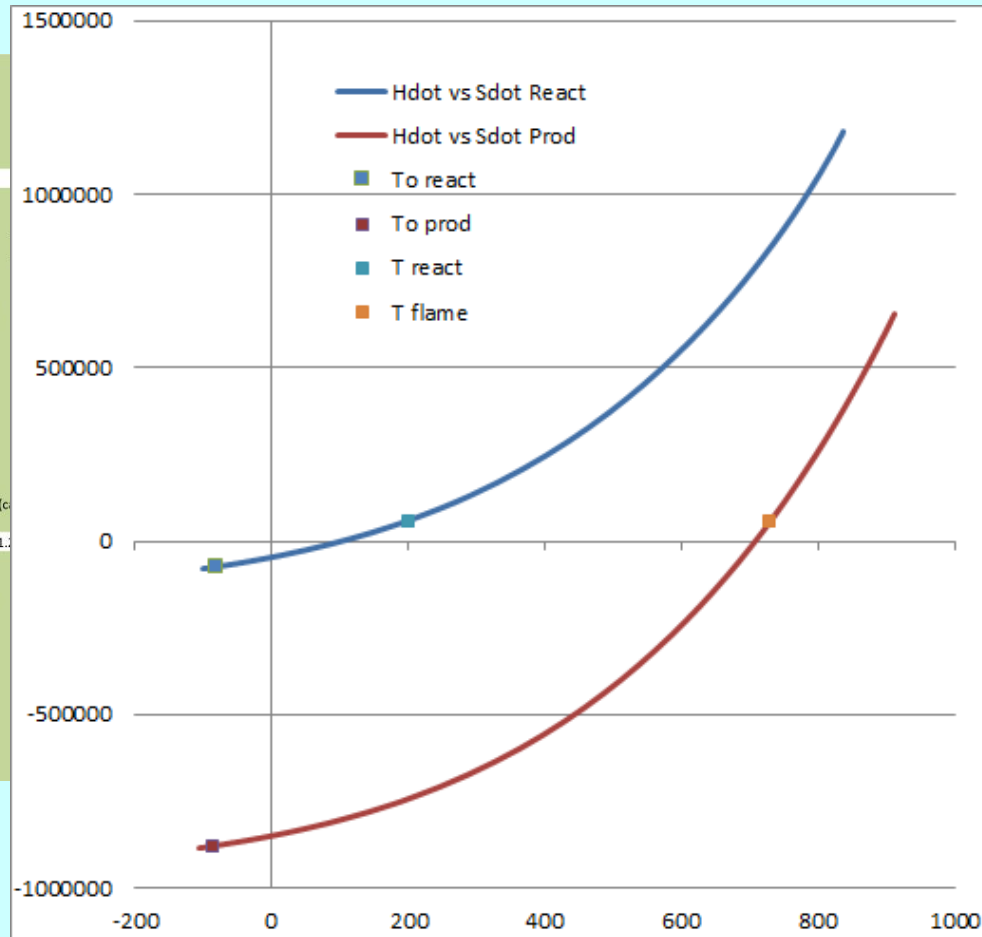
Adapted from Table 20.2 of Gyftopoulos, Beretta, Thermodynamics. Foundations and applications, Dover, 2005

# Adiabatic flame temperature calculations



Steady-state constant-pressure burner

lambda	1	To	298.15
nitrogen?	1	Treactants	700
C	1	T	2666.064
H	4		
M	16.043	16	
	nu	n react	n prod
CH4	-1	1	0
N2	0	7.54	7.54
O2	-2	2	0
CO2	1	0	1
H2O	2	0	2
n_tot		10.54	10.54
		Coefficients from Table B5	
	nu	n react	a
N2	0	7.54	72
O2	-2	0	10.3
CO2	1	1	-55.6
H2O	2	2	180
n_tot		10.54	
		Coefficients from Table B4 (c)	
	nu	n react	a
CH4	-1	1	19.3
			5.21E-02
			1.1
			18814.56
			38.623
		Coefficients from Table B5	
	nu	n react	a
N2	0	7.54	72
O2	-2	2	10.3
CO2	1	0	-55.6
H2O	2	0	180
n_tot		10.54	
			114436.03
			241.711

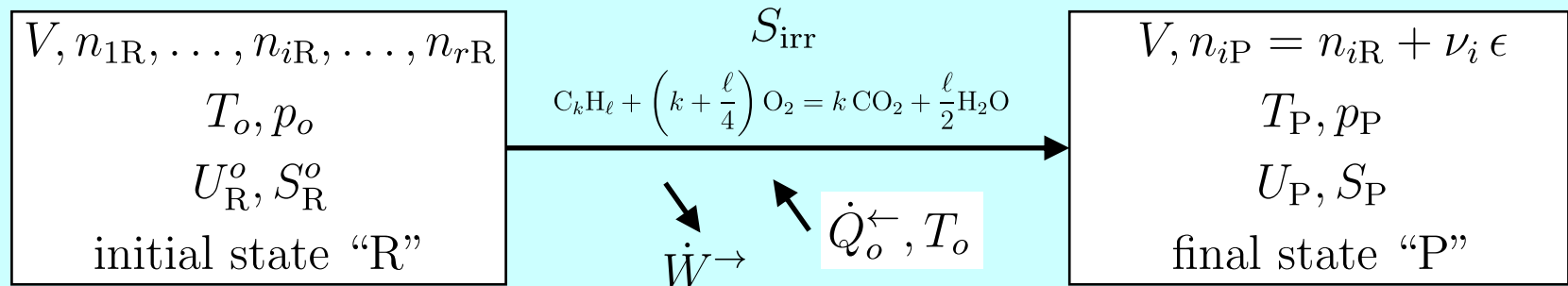


Products					
S dot	H dot	T	S dot	H dot	
0	-99.97376	-80498.806	280	-105.8162	-883555.49
5	-80.6	-74900	298.15	-85.9	-877800.00
0	-30.55922	-58713.829	350	-34.66205	-861227.84
0	11.861578	-42826.538	400	8.5517546	-845044.39
0	49.939158	-26659.489	450	47.172405	-828647.42
0	84.594441	-10210.971	500	82.184321	-812030.18
0	116.47891	6517.92924	550	114.28069	-795190.57
0	146.07025	23524.221	600	143.96813	-778129.45
0	173.72906	40803.5574	650	171.6283	-760849.62
0	199.73412	58350.5834	700	197.55601	-743355.11
0	247.61928	94222.519	800	245.09868	-707741.89
0	291.02464	131083.893	900	287.9727	-671333.47
0	330.82903	168870.958	1000	327.1147	-634176.57
0	367.6533	207513.699	1100	363.1925	-596318.14
0	401.9509	246936.449	1200	396.70019	-557804.00
0	434.06189	287058.143	1300	428.0144	-518678.18
0	464.24662	327792.396	1400	457.42945	-478982.63
0	492.70781	369047.486	1500	485.18011	-438757.19
0	519.6054	410726.297	1600	511.45701	-398039.58
0	545.06689	452726.237	1700	536.4173	-356865.49
0	569.19469	494939.151	1800	560.19226	-315268.70
0	592.07156	537251.233	1900	582.89283	-273281.19
0	613.76455	579542.945	2000	604.61371	-230933.27
0	634.32806	621688.935	2100	625.43653	-188253.68
0	653.80616	663557.961	2200	645.43215	-145269.70
0	672.23438	705012.831	2300	664.66262	-102007.29
0	689.64115	745910.336	2400	683.18256	-58491.14
0	706.04895	786101.193	2500	701.04039	-14744.78
0	721.47519	825429.998	2600	718.27926	29209.34
0	735.93301	863735.178	2700	734.9378	73349.80
0	749.43182	900848.947	2800	751.05079	117656.08
0	761.97787	936597.266	2900	766.64961	162108.58
0	773.57463	970799.814	3000	781.76278	206688.50
0	784.22317	1003269.95	3100	796.41624	251377.81
0	793.92242	1033814.68	3200	810.63368	296159.21
0	802.66947	1062234.65	3300	824.43683	341016.07

E18-AdiabaticFlameTemperature(withEquilibrium).xlsx



# Adiabatic flame temperature and pressure in a closed bomb



**Volume balance** (assuming GD mix of ideal gases):  $V_P = \frac{n_P R T_P}{p_P} = V_R = \frac{n_R R T_o}{p_o}$

**Heating value** setup:  $W^{\rightarrow} = 0, \epsilon = n_{C_k H_l}, n_P = n_R + \nu \epsilon, \frac{p_P}{p_o} = 1 + \nu \frac{\epsilon}{n_R}$

$$U_P - U_R^o = -Q_o^{\rightarrow} = -\epsilon HV_{\text{actual}}^{\text{bomb}} \quad \text{where} \quad HV_{\text{actual}}^{\text{bomb}} = -\Delta u^o + \frac{n_{H_2O,f}}{n_{C_k H_l}} u_{H_2O,fg}(T_o)$$

**Adiabatic flame temperature** setup:  $T_P = T_{\text{ad}}, Q_o^{\rightarrow} = 0, W^{\rightarrow} = 0, \frac{p_{\text{ad}}}{p_o} = \frac{T_{\text{ad}}}{T_o} \left(1 + \nu \frac{\epsilon}{n_R}\right)$

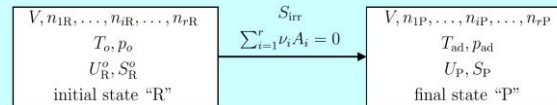
$$0 = U_P(T_{\text{ad}}) - U_R^o = U_P(T_{\text{ad}}) - U_P^o + U_P^o - U_R^o = n_P \int_{T_o}^{T_{\text{ad}}} \sum_i y_{iP} c_{v,ii}(T') dT' - \epsilon HV_{\text{actual}}^{\text{bomb}} \Rightarrow \text{solve for } T_{\text{ad}}$$

$$\begin{aligned} \dot{S}_{irr} &= S_P(T_{\text{ad}}, p_{\text{ad}}) - S_R^o = S_P(T_{\text{ad}}, p_{\text{ad}}) - S_P^o + S_P^o - S_R^o \\ &= n_P \int_{T_o}^{T_{\text{ad}}} \frac{\sum_i y_{iP} c_{p,ii}(T')}{T'} dT' - n_P R \ln \frac{p_{\text{ad}}}{p_o} + \epsilon \Delta s^o \approx -(0.22 \div 0.26) \epsilon \frac{\Delta g^o}{T_o} \end{aligned}$$

Typical values of  $\frac{T_o \dot{S}_{irr}}{-\epsilon \Delta g^o}$  are in the range 22% to 26%

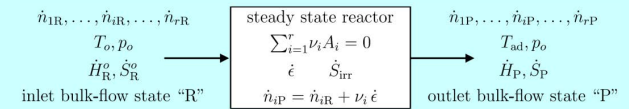
# Adiabatic flame temperature

Closed fixed-volume  
combustion chamber



Fuel	Formula	$T_b$ K	$p_b$ atm	$\frac{T_o S_{irr}}{(-n_{1a} \Delta g^o)}$ %
Hydrogen	H <sub>2</sub>	2870.8	8.33	17.2
Carbon	C	2677.9	7.58	22.7
Methane	CH <sub>4</sub>	2655.4	9.02	24.2
Acetylene	C <sub>2</sub> H <sub>2</sub>	3009.7	10.00	19.2
Ethylene	C <sub>2</sub> H <sub>4</sub>	2811.0	9.63	22.4
Ethane	C <sub>2</sub> H <sub>6</sub>	2691.1	9.42	24.9
Propylene	C <sub>3</sub> H <sub>6</sub>	2770.7	9.69	23.7
Propane	C <sub>3</sub> H <sub>8</sub>	2698.7	9.56	25.3
n-Butane	C <sub>4</sub> H <sub>10</sub>	2702.7	9.64	25.6
n-Pentane	C <sub>5</sub> H <sub>12</sub>	2704.0	9.68	25.7
Benzene	C <sub>6</sub> H <sub>6</sub>	2767.6	9.61	23.7
n-Hexane	C <sub>6</sub> H <sub>14</sub>	2703.9	9.71	25.8
n-Heptane	C <sub>7</sub> H <sub>16</sub>	2707.6	9.75	25.9
n-Octane	C <sub>8</sub> H <sub>18</sub>	2706.9	9.76	26.0
Isooctane	C <sub>8</sub> H <sub>18</sub>	2702.6	9.74	26.2
n-Nonane	C <sub>9</sub> H <sub>20</sub>	2707.1	9.77	26.0
n-Decane	C <sub>10</sub> H <sub>22</sub>	2707.2	9.78	26.1

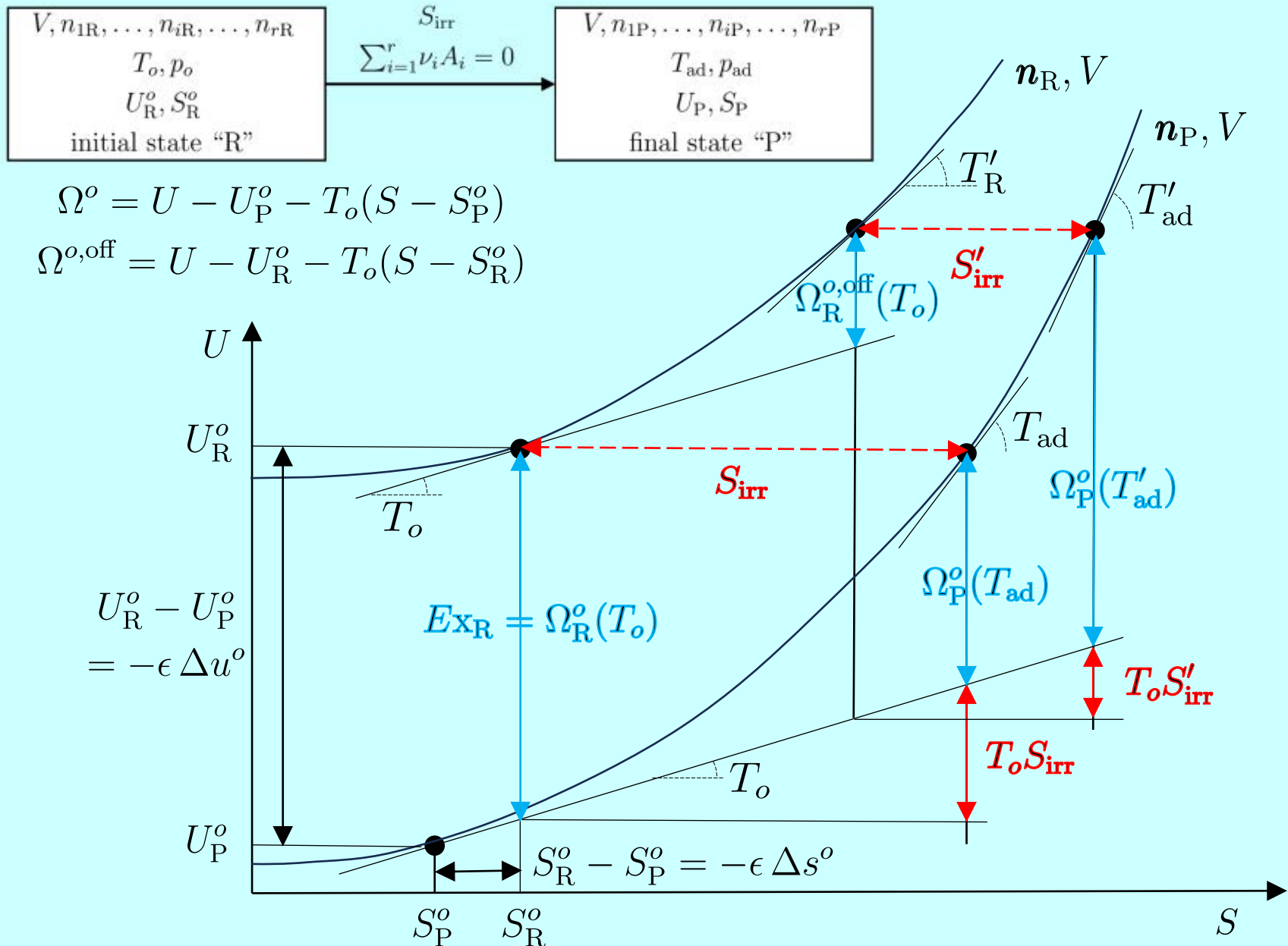
Steady-state constant-  
pressure burner



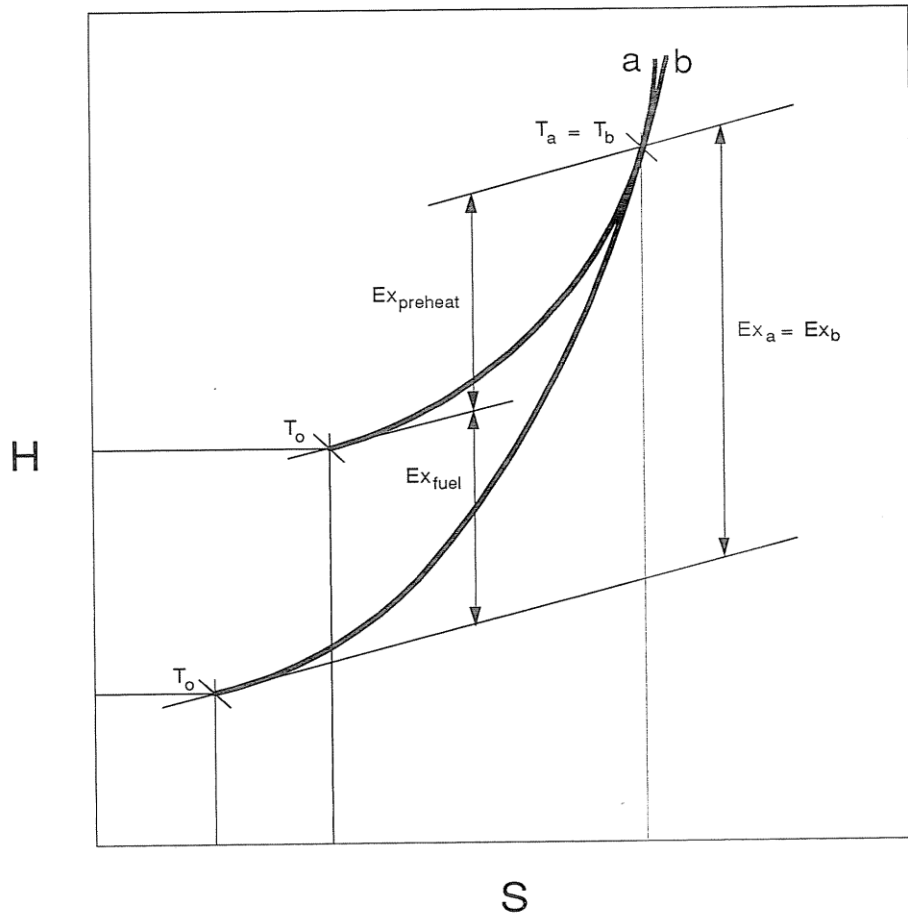
$T_b$ K	$\frac{T_o \dot{S}_{irr}}{(-\dot{n}_{1a} \Delta g^o)}$ %
2448.5	20.9
2326.0	26.0
2266.0	28.3
2598.0	22.6
2416.6	26.2
2300.5	29.0
2378.5	27.6
2307.9	29.4
2311.8	29.6
2313.2	29.7
2382.6	27.5
2313.2	29.8
2316.7	29.9
2316.1	30.0
2312.1	30.2
2316.3	30.0
2316.4	30.1

Adapted from Figures 31.8 and 31.9 of Gyftopoulos, Beretta, Thermodynamics. Foundations and applications, Dover, 2005

# Summary on an energy vs entropy diagram

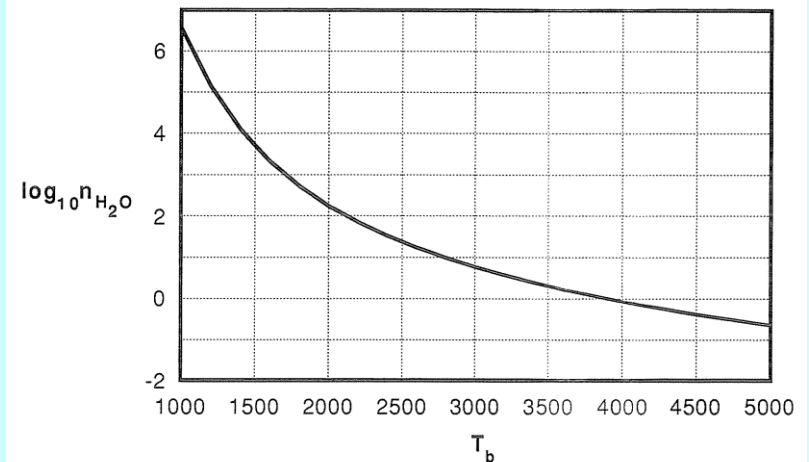


# Is a reversible combustion possible?



For  $n_{H_2} = 1 \text{ mol}$ ,  $n_{O_2} = 1/2 \text{ mol}$  and  $p_b = p_o$  we find

$$n_{H_2O} \left( \frac{3}{2} + n_{H_2O} \right)^{1/2} = \left( \frac{1}{2} \right)^{-1/2} \exp \left( -6.866 + \frac{29911}{T_b} \right)$$



Figures 3 and 4 from G.P. Beretta, A.M. Lezzi, A. Niro, and M. Silvestri, [On the concept of a reversible flame](#), in Energy for the Transition Age, FLOWERS '92, Edited by S.S. Stecco and M.J. Moran, Nova Science Pu. Inc., New York, Additional Proceedings, pp. 165-177 (1992).

The question was first raised and discussed in Keenan (1941) and Obert (1948,1973), and later also by Richter & Knoche (1983), Dunbar & Lior (1994).

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**Slide 28:**

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## 2.43 Advanced Thermodynamics

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