

Lecture 3 - 2003

Kollbruner Section 5.2 Characteristics of Thin Walled Sections and .. Kollbruner Section 5.3 Bending without Twist

thin walled => (cross section shape arbitrary and thickness can vary)

axial stresses and shear stress along center of wall govern

normal (to curved cross section) stress neglected

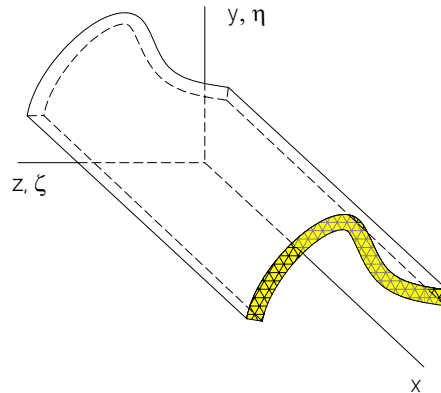
position determined by curvilinear coordinate s along center line of cross section

St. Venant torsion not a player as $K \sim t^3$

use shear flow:

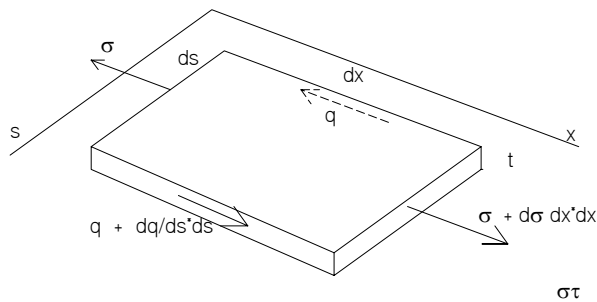
$$z \Rightarrow \zeta$$

$$x \Rightarrow \xi$$



a) equilibrium of wall element:

$$\tau = \tau_{XS} = \tau_{SX}$$



$$\left[\sigma + \left(\frac{d}{dx} \sigma \right) \cdot dx \right] \cdot t \cdot ds - \sigma \cdot t \cdot ds + \left(q + \frac{d}{ds} q \cdot ds \right) \cdot dx - q \cdot dx = 0$$

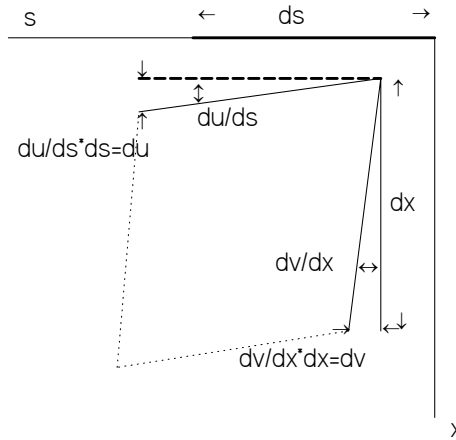
using $q = \tau \cdot t$
as it includes $\tau(s)$ and $t(s)$

$$\frac{d}{ds} q + \left(\frac{d}{dx} \sigma \right) \cdot t = 0$$

b) compatibility (shear strain)

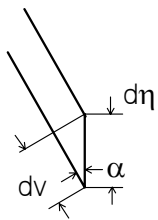
$$\frac{d}{ds}u + \frac{d}{dx}v = \gamma$$

v is displacement in s direction
u is displacement in x direction

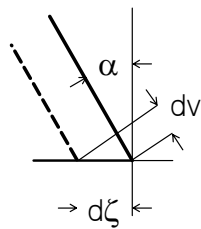


view looking \perp to surface

c) tangential displacement (δv) in terms of η , ζ and ϕ



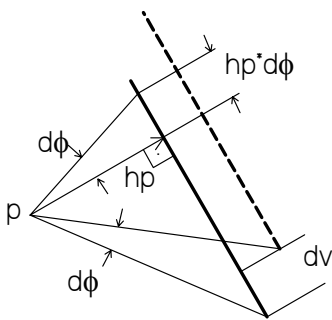
y component



z component

v is displacement in s direction
 η is displacement in y direction
 ζ is displacement in z direction
and ϕ is rotation
all at point s on cross section.

δv is differential over distance dx
 $\delta \eta$ is component of δv in y
 $\delta \zeta$ is component of δv in z
direction
and $h_p \cdot \delta \phi$ is component due to
differential rotation between x and
 $x+dx$



rotation component

superposition =>

$$\delta v = \delta \eta \cdot \cos(\alpha) + \delta \zeta \cdot \sin(\alpha) + h_p \cdot \delta \phi$$

η , ζ and ϕ depend on s and x while α and h_p are independent of x (prismatic section)
rewrite as:

$$\frac{\delta v}{\delta x} = \frac{\delta \eta}{\delta x} \cdot \cos(\alpha) + \frac{\delta \zeta}{\delta x} \cdot \sin(\alpha) + h_p \cdot \frac{\delta \phi}{\delta x}$$

further assumptions:

1) preservation of cross section shape $\Rightarrow \zeta = \zeta(x); \eta = \eta(x) \phi = \phi(x)$

2) shear though finite is small $\sim 0 \Rightarrow \frac{d}{ds} u = -\left(\frac{d}{dx} v\right)$

3) Hooke's law holds $\Rightarrow \sigma = E \cdot \frac{\delta u}{\delta x}$ axial stress

equilibrium for the cross section:

$$\int \sigma \, dA = N_x$$

$$\int \sigma \cdot y \, dA = -M_z$$

$$\int \sigma \cdot z \, dA = M_y$$

$$\int \tau \cdot h_p \, dA = \int q \cdot h_p \, ds = T_p$$

$$\int \tau \cdot \cos(\alpha) \, dA = \int q \cdot \cos(\alpha) \, ds = V_y$$

$$\int \tau \cdot \sin(\alpha) \, dA = \int q \cdot \sin(\alpha) \, ds = V_z$$

N_x = axial force
 M_z and M_y are bending moments wrt y and z respectively
 note integral expressions
 T_p is torsional moment wrt cross sectional point P
 Q_x and Q_y shear forces

5.3 Bending without twist

$$\int \sigma \, dA = 0$$

$$\int q \cdot h_p \, ds = 0$$

possible only if lateral loads pass through P

$$\frac{\delta v}{\delta x} = \frac{\delta \eta}{\delta x} \cdot \cos(\alpha) + \frac{\delta \zeta}{\delta x} \cdot \sin(\alpha) + h_p \cdot \frac{\delta \phi}{\delta x} \text{ from above,}$$

$$\text{using } \frac{d}{ds} u = - \left(\frac{d}{dx} v \right) \text{ with no twist } \Rightarrow \frac{\delta \phi}{\delta x} = 0$$

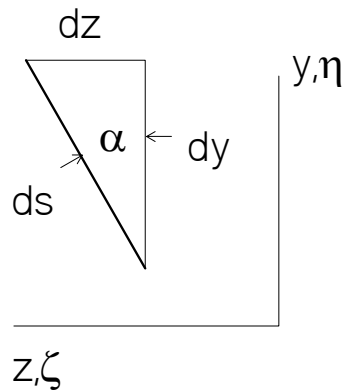
becomes: $\frac{\delta u}{\delta s} = - \frac{d\eta}{dx} \cdot \cos(\alpha) - \frac{d\zeta}{dx} \cdot \sin(\alpha)$ which can be integrated to become:

$$u = -\eta' \cdot \int \cos(\alpha) ds - \zeta' \cdot \int \sin(\alpha) ds + u_0(x) \text{ where } \frac{d\zeta}{dx} = \zeta' \text{ (prime is control F7)}$$

and $\frac{\delta \eta}{\delta x}$ is η' , and comes outside the integral due to our prismatic assumption

$$dY (=dy) = ds \cdot \cos(\alpha), dZ (=dz) = ds \cdot \sin(\alpha),$$

where Y and Z refer to a coordinate system with an arbitrary origin, whereas x and y are defined centroidal (refer to center of area) =>



$$u(x) = -\eta' \cdot \int 1 dY - \zeta' \cdot \int 1 dZ + u_0(x)$$

$$u = -\eta' \cdot Y - \zeta' \cdot Z + u_0(x)$$

which says longitudinal displacement u is distributed linearly across cross section (plane sections remain plane)

u_0 is the constant of integration which is $f(x)$

axial strain = $du/dx \Rightarrow u' = -\eta'' \cdot Y - \zeta'' \cdot Z + u'_0(z)$ and

$$\sigma = E \cdot u' = -E \cdot \eta'' \cdot Y - \zeta'' \cdot Z + E \cdot u'_0(z)$$

$$\text{now with } \int \sigma dA = 0$$

$$\int \sigma dA = \int E \cdot u' da = - \int E \cdot \eta'' \cdot Y dA - \int E \cdot \zeta'' \cdot Z dA + \int E \cdot u'_0(z) dA = 0 \Rightarrow$$

$$-E \cdot \eta'' \cdot \int Y dA - E \cdot \zeta'' \cdot \int Z dA + E \cdot u'_0(z) \cdot \int 1 dA = 0 \Rightarrow$$

$$E \cdot u'_0(z) = E \cdot \eta'' \cdot \frac{\int Y dA}{A} + E \cdot \zeta'' \cdot \int Z dA \quad \text{determines } E \cdot u'_0(z) \text{ in stress } \sigma \Rightarrow$$

$$\sigma = -E \cdot \eta'' \cdot Y - E \cdot \zeta'' \cdot Z + E \cdot \eta'' \cdot \frac{\int Y dA}{A} + E \cdot \zeta'' \cdot \frac{\int Z dA}{A} \quad \text{rearranged becomes}$$

$$\sigma = -E \cdot \eta'' \cdot \left(Y - \frac{\int Y dA}{A} \right) - E \cdot \zeta'' \cdot \left(Z - \frac{\int Z dA}{A} \right) \quad \text{but}$$

$$\frac{\int Y dA}{A} \text{ is the definition of the } y \text{ position of the centroid and } \frac{\int Z dA}{A} \text{ the } z \text{ position } \Rightarrow$$

$$y = Y - \frac{\int Y dA}{A} \quad \text{and} \quad z = Z - \frac{\int Z dA}{A} \quad \text{and } \sigma = -E \cdot \eta'' \cdot y - E \cdot \zeta'' \cdot z$$

where y and z are the position of the points in the centroidal coordinate system.

η'' and ζ'' are determined by the equilibrium conditions

$$\int \sigma \cdot y dA = -M_z \quad \int \sigma \cdot z dA = M_y \quad \sigma = -E \cdot \eta'' \cdot y - E \cdot \zeta'' \cdot z$$

$$\int \sigma \cdot y dA = \int (-E \cdot \eta'' \cdot y - E \cdot \zeta'' \cdot z) \cdot y dA = -E \cdot \eta'' \cdot \left(\int y \cdot y dA \right) - E \cdot \zeta'' \cdot \left(\int z \cdot y dA \right) = -M_z$$

$$\int \sigma \cdot z \, dA = \int (-E \cdot \eta'' \cdot y - E \cdot \zeta'' \cdot z) \cdot z \, dA = -E \cdot \eta'' \int y \cdot z \, dA - E \cdot \zeta'' \int z \cdot z \, dA = M_y$$

noting that $\int z \cdot z \, dA = I_y$, $\int z \cdot y \, dA = I_{yz}$ and $\int y \cdot y \, dA = I_z$ and solving the two equations for two unknowns $E \cdot \eta''$ and $E \cdot \zeta''$ leads to =>

Given

$$-E \eta'' \cdot I_z - E \zeta'' \cdot I_{yz} = -M_z$$

$$-E \eta'' \cdot I_{yz} - E \zeta'' \cdot I_y = M_y$$

solving two equations two unknowns

$$\text{Find}(E \eta'', E \zeta'') \rightarrow \begin{pmatrix} \frac{I_{yz} \cdot M_y + M_z \cdot I_y}{-I_{yz}^2 + I_y \cdot I_z} \\ \frac{-I_{yz} \cdot M_z - M_y \cdot I_z}{-I_{yz}^2 + I_y \cdot I_z} \end{pmatrix}$$

$$E \eta'' := \frac{(I_{yz} \cdot M_y + M_z \cdot I_y)}{(-I_{yz}^2 + I_y \cdot I_z)}$$

$$E \zeta'' := \frac{(-I_{yz} \cdot M_z - M_y \cdot I_z)}{(-I_{yz}^2 + I_y \cdot I_z)}$$

reversed signs

now substituting back into the relation for axial stress ($\sigma := -E \cdot \eta'' \cdot y - E \cdot \zeta'' \cdot z$) =>

$$\sigma := -E \eta'' \cdot y - E \zeta'' \cdot z$$

$$\sigma \rightarrow \frac{-(I_{yz} \cdot M_y + M_z \cdot I_y)}{-I_{yz}^2 + I_y \cdot I_z} \cdot y - \frac{-I_{yz} \cdot M_z - M_y \cdot I_z}{-I_{yz}^2 + I_y \cdot I_z} \cdot z$$

or rearranging in terms of M_y and M_z =>

$$\sigma := \frac{(-I_y \cdot y + I_{yz} \cdot z) \cdot M_z + (I_z \cdot z - I_{yz} \cdot y) \cdot M_y}{(-I_{yz}^2 + I_y \cdot I_z)}$$

as a check on this development let:

$$I_{yz} := 0 \quad \text{and} \quad \sigma := \frac{-I_{yz} \cdot M_y + M_z \cdot I_y}{(-I_{yz}^2 + I_y \cdot I_z)} \cdot y - \frac{(-I_{yz} \cdot M_z - M_y \cdot I_z)}{(-I_{yz}^2 + I_y \cdot I_z)} \cdot z$$

$$\sigma \rightarrow \frac{-M_z}{I_z} \cdot y + \frac{M_y}{I_y} \cdot z \quad \text{which matches our previous understanding on bending}$$

c) Shear Stress

integration of $\frac{d}{ds} q + \left(\frac{d}{dx} \sigma \right) \cdot t = 0$ (the equilibrium relationship above) along s leads to :

$$q(s, x) = q_1(x) - \int_0^s \left(\frac{d}{dx} \sigma \right) \cdot t \, ds$$

where $q_1(x)$ is $f(x)$ and represents the shear flow

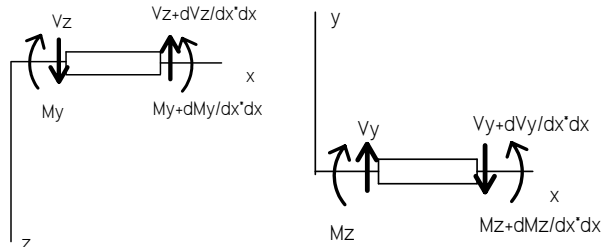
at the start of the region. it is 0 at a stress free boundary which is convenient for an open section:

$$q_1(x) = 0$$

$$\frac{d}{dx} \sigma = \frac{d}{dx} \left[\frac{-I_{yz} \cdot M_y + M_z \cdot I_y}{(-I_{yz}^2 + I_y \cdot I_z)} \cdot y - \frac{(-I_{yz} \cdot M_z - M_y \cdot I_z)}{(-I_{yz}^2 + I_y \cdot I_z)} \cdot z \right]$$

where only M_y and M_z are x dependent and

$$\frac{d}{dx} M_z = V_y \quad \text{and} \quad \frac{d}{dx} M_y = -V_z \Rightarrow$$



$$M_z(x) := V_y \cdot x \quad M_y(x) := -V_z \cdot x \quad I_{yz} := I_{yz}$$

$$\frac{d}{dx} \left[\frac{-I_{yz} \cdot M_y(x) + M_z(x) \cdot I_y}{(-I_{yz}^2 + I_y \cdot I_z)} \cdot y - \frac{(-I_{yz} \cdot M_z(x) - M_y(x) \cdot I_z)}{(-I_{yz}^2 + I_y \cdot I_z)} \cdot z \right] \rightarrow \frac{I_{yz} \cdot V_z - V_y \cdot I_y}{-I_{yz}^2 + I_y \cdot I_z} \cdot y - \frac{-I_{yz} \cdot V_y + V_z \cdot I_z}{-I_{yz}^2 + I_y \cdot I_z} \cdot z$$

copy and substitute

$$q(s, x) = - \int_0^s \left(\frac{d}{dx} \sigma \right) \cdot t \, ds = - \int_0^s \left[\frac{(I_{yz} \cdot V_z - V_y \cdot I_y)}{(-I_{yz}^2 + I_y \cdot I_z)} \cdot y - \frac{(-I_{yz} \cdot V_y + V_z \cdot I_z)}{(-I_{yz}^2 + I_y \cdot I_z)} \cdot z \right] \cdot t \, ds$$

$$q(s, x) = \frac{-1}{-I_{yz}^2 + I_y \cdot I_z} \cdot \left[(I_{yz} \cdot V_z - V_y \cdot I_y) \cdot \int_0^s y \cdot t \, ds - (-I_{yz} \cdot V_y + V_z \cdot I_z) \cdot \int_0^s z \cdot t \, ds \right]$$

if we designate the integrals which are the static moments of the cross section area: Q_y and Q_z :

$$Q_z = \int_0^s y \cdot t \, ds \quad Q_y = \int_0^s z \cdot t \, ds$$

$$q(s, x) := \frac{-1}{-I_{yz}^2 + I_y \cdot I_z} \cdot \left[(I_{yz} \cdot V_z - V_y \cdot I_y) \cdot Q_z - (-I_{yz} \cdot V_y + V_z \cdot I_z) \cdot Q_y \right] \quad \text{or rearranging as we did for axial stress}$$

$$q(s, x) \text{ collect, } V_y, V_z \rightarrow \frac{-1}{-I_{yz}^2 + I_y \cdot I_z} \cdot (-I_y \cdot Q_z + I_{yz} \cdot Q_y) \cdot V_y - \frac{1}{-I_{yz}^2 + I_y \cdot I_z} \cdot (I_{yz} \cdot Q_z - I_z \cdot Q_y) \cdot V_z$$

$$q(s, x) := \frac{-1}{(-I_{yz}^2 + I_y \cdot I_z)} \cdot \left[(I_{yz} \cdot Q_z - I_z \cdot Q_y) \cdot V_z + (-I_y \cdot Q_z + I_{yz} \cdot Q_y) \cdot V_y \right]$$

or if the axes are principal ($I_{xy} = 0$) $I_{yz} := 0$

$$q(s, x) := \frac{-1}{(-I_{yz}^2 + I_y \cdot I_z)} \cdot \left[(I_{yz} \cdot Q_z - I_z \cdot Q_y) \cdot V_z + (-I_y \cdot Q_z + I_{yz} \cdot Q_y) \cdot V_y \right]$$

$$q(s, x) \text{ simplify collect, } Q_y \rightarrow \frac{V_z}{I_y} \cdot Q_y + Q_z \cdot \frac{V_y}{I_z} \quad q(s, x) := \left(\frac{Q_y \cdot V_z}{I_y} + \frac{Q_z \cdot V_y}{I_z} \right) \quad I_{yz} := 0$$



d) Shear Center

the above relationships apply for bending without twist i.e. when

$$\int \sigma \, dA = 0 \quad \int q \cdot h_p \, ds = 0 \quad \text{possible only if lateral loads pass through P}$$

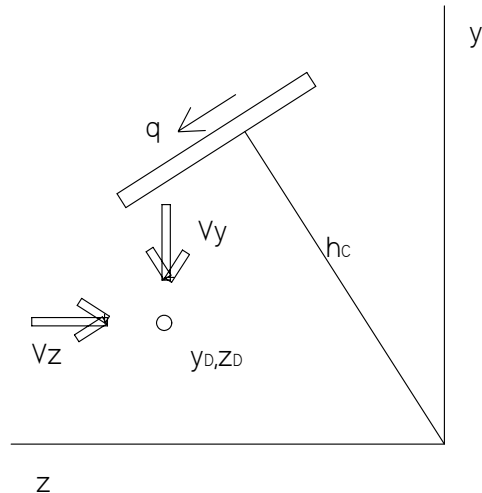
$$q \rightarrow \frac{-1}{-I_{zy}^2 + I_y \cdot I_z} \cdot \left[(I_{zy} \cdot Q_z - I_z \cdot Q_y) \cdot V_z + (-I_y \cdot Q_z + I_{zy} \cdot Q_y) \cdot V_y \right] \quad \text{from above (reset in hidden area)}$$

this point P was designated (by Maillart in 1921, see page 106 Kollbrunner) as the shear center. In the centroidal coordinate system, this is located at y_D and z_D . the second condition applies for a shear force V. The center of action must pass through P. Divide Q into components V_y and V_z

thus from (Q_y) , the moment the moment equilibrium wrt C (in centroidal coordinates) is:

$$\int q(V_y) \cdot h_c \, ds + V_y \cdot z_D = 0$$

$$\int q(V_y) \cdot h_c \, ds = -V_y \cdot z_D$$



$q(V_y)$ is q with V_z set to 0 and h_c is perpendicular distance from centroid to line of action (i.e. $y(s)$ and $z(s)$).

set $V_z := 0$ reset $q := \frac{-1}{(-I_{yz}^2 + I_y \cdot I_z)} \cdot [(I_{yz} \cdot Q_z - I_z \cdot Q_y) \cdot V_z + (-I_y \cdot Q_z + I_{yz} \cdot Q_y) \cdot V_y]$

$q \rightarrow \frac{-1}{-I_{zy}^2 + I_y \cdot I_z} \cdot (-I_y \cdot Q_z + I_{zy} \cdot Q_y) \cdot V_y$ substitute

$$\int q(V_y) \cdot h_c \, ds = -V_y \cdot z_D = \int \frac{-1}{(-I_{zy}^2 + I_y \cdot I_z)} \cdot (-I_y \cdot Q_z + I_{zy} \cdot Q_y) \cdot V_y \cdot h_c \, ds$$

$$-V_y \cdot z_D = \frac{-V_y}{(-I_{zy}^2 + I_y \cdot I_z)} \left[\int (-I_y \cdot Q_z + I_{zy} \cdot Q_y) \cdot h_c \, ds \right] \quad \text{or ...}$$

$$z_D = \frac{1}{(-I_{zy}^2 + I_y \cdot I_z)} \left[\int (-I_y \cdot Q_z + I_{zy} \cdot Q_y) \cdot h_c \, ds \right] = \frac{-I_y \cdot \int Q_z \cdot h_c \, ds + I_{zy} \cdot \int Q_y \cdot h_c \, ds}{-I_{zy}^2 + I_y \cdot I_z}$$

moment from V_z $\int q(V_z) \cdot h_c \, ds - V_z \cdot y_D = 0$ $\int q(V_z) \cdot h_c \, ds = V_z \cdot y_D$

$q(V_z)$ is q with V_y set to 0 and h_c is perpendicular distance from centroid to line of action (i.e. $y(s)$ and $z(s)$).

reset

$$\text{reset } V_z := V_z \quad \text{set } V_y := 0 \quad q := \frac{-1}{(-I_{yz}^2 + I_y \cdot I_z)} \cdot [(I_{yz} \cdot Q_z - I_z \cdot Q_y) \cdot V_z + (-I_y \cdot Q_z + I_{yz} \cdot Q_y) \cdot V_y]$$

$$q \rightarrow \frac{-1}{-I_{zy}^2 + I_y \cdot I_z} \cdot (I_{zy} \cdot Q_z - I_z \cdot Q_y) \cdot V_z \quad \text{substitute}$$

$$\int q(V_z) \cdot h_c \, ds = V_z \cdot y_D = \int \frac{-1}{(-I_{zy}^2 + I_y \cdot I_z)} \cdot (I_{zy} \cdot Q_z - I_z \cdot Q_y) \cdot V_z \cdot h_c \, ds$$

$$V_z \cdot y_D = \int \frac{-1}{(-I_{zy}^2 + I_y \cdot I_z)} \cdot (I_{zy} \cdot Q_z - I_z \cdot Q_y) \cdot V_z \cdot h_c \, ds = \frac{-V_z}{(-I_{zy}^2 + I_y \cdot I_z)} \cdot \left(I_{zy} \cdot \int Q_z \cdot h_c \, ds - I_z \cdot \int Q_y \cdot h_c \, ds \right)$$

$$y_D = \frac{-1}{(-I_{zy}^2 + I_y \cdot I_z)} \cdot \left(I_{zy} \cdot \int Q_z \cdot h_c \, ds - I_z \cdot \int Q_y \cdot h_c \, ds \right)$$

for principal axes ($I_{yz} = 0$): $I_{zy} := 0$

$$y_D := \frac{-1}{(-I_{zy}^2 + I_y \cdot I_z)} \cdot \left(I_{zy} \cdot \int Q_z \cdot h_c \, ds - I_z \cdot \int Q_y \cdot h_c \, ds \right) \quad z_D := \frac{-I_y \cdot \int Q_z \cdot h_c \, ds + I_{zy} \cdot \int Q_y \cdot h_c \, ds}{-I_{zy}^2 + I_y \cdot I_z}$$

$$y_D = \frac{\int Q_y \cdot h_c \, ds}{I_y}$$

$$z_D = \frac{-\int Q_z \cdot h_c \, ds}{I_z}$$